

# The Fluorescence Detector of the Pierre Auger Observatory

## The Pierre Auger Collaboration

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United Kingdom
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## Abstract

The Pierre Auger Observatory is a hybrid detector for ultra-high energy cosmic rays. It combines a surface array to measure secondary particles at ground level together with a fluorescence detector to measure the development of air showers in the atmosphere above the array. The fluorescence detector comprises 24 large telescopes specialized for measuring the nitrogen fluorescence caused by charged particles of cosmic ray air showers. In this paper we describe the components of the fluorescence detector including its optical system, the design of the camera, the electronics, and the systems for relative and absolute calibration. We also discuss the operation and the monitoring of the detector. Finally, we evaluate the detector performance and precision of shower reconstructions.

*Key words:* cosmic rays, fluorescence detector

*PACS:* 96.40.-z, 96.40.Pq, 98.70.Sa

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## 1. Introduction

The hybrid detector of the Pierre Auger Observatory [1] consists of 1600 surface stations – water Cherenkov tanks and their associated electronics – and 24 air fluorescence telescopes. The Observatory is located outside the city of Malargüe, Argentina (69° W, 35° S, 1400 m a.s.l.) and the detector layout is shown in Fig. 1. Details of the construction, deployment and maintenance of the array of surface detectors are described elsewhere [2]. In this paper we will concentrate on details of the fluorescence detector and its performance.

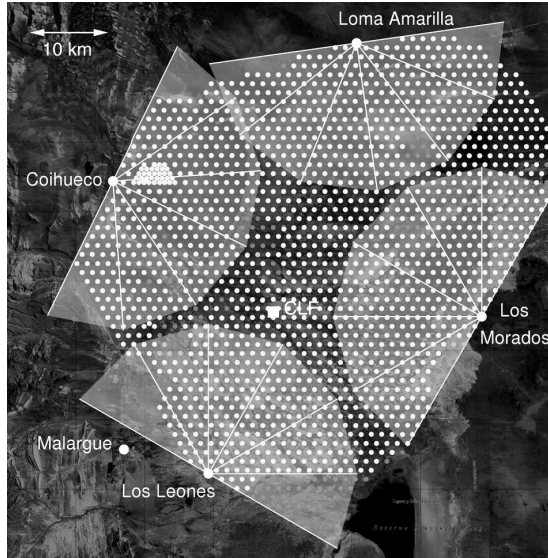


Figure 1: Status of the Pierre Auger Observatory as of March 2009. Gray dots show the positions of surface detector stations, lighter gray shades indicate deployed detectors, while dark gray defines empty positions. Light gray segments indicate the fields of view of 24 fluorescence telescopes which are located in four buildings on the perimeter of the surface array. Also shown is a partially completed infill array near the Coihueco station and the position of the Central Laser Facility (CLF, indicated by a white square). The description of the CLF and also the description of all other atmospheric monitoring instruments of the Pierre Auger Observatory is available in [3].

The detection of ultra-high energy<sup>1</sup> cosmic rays using nitrogen fluorescence emission induced by extensive air showers is a well established technique, used previously by the Fly’s Eye [4] and HiRes [5] experiments. It is used also for the Telescope Array [6] project that is currently under construction, and it has been proposed for the satellite-based EUSO (Extreme Universe Space Observatory) and OWL (Orbiting Wide-angle Light-collectors) projects, which have recently evolved into the proposals of the JEM-EUSO mission [7] (EUSO onboard the

<sup>1</sup>For the purposes of this paper we define the ultra-high energy to be  $\gtrsim 10^{18}$  eV.



Japanese Experiment Module at the International Space Station), and of the S-EUSO free-flying satellite mission [8].

Charged particles generated during the development of extensive air showers excite atmospheric nitrogen molecules, and these molecules then emit fluorescence light in the  $\sim 300 - 430$  nm range (see Figure 2). The number of emitted fluorescence photons is proportional to the energy deposited in the atmosphere due to electromagnetic energy losses by the charged particles. By measuring the rate of fluorescence emission as a function of atmospheric slant depth  $X$ , an air fluorescence detector measures the *longitudinal development profile*  $\frac{dE}{dX}(X)$  of the air shower. The integral of this profile gives the total energy dissipated electromagnetically, which is approximately 90% of the total energy of the primary cosmic ray.

For any waveband, the fluorescence yield is defined as the number of photons emitted in that band per unit of energy loss by charged particles. The absolute fluorescence yield in air at 293 K and 1013 hPa from the 337 nm fluorescence band is  $5.05 \pm 0.71$  photons/MeV of energy deposited, as measured in [9]. For the reconstruction of the cosmic ray showers at the Pierre Auger Observatory, this absolute measurement is combined with the relative yields at other fluorescence bands as measured in [10].<sup>2</sup> Since a typical cosmic ray shower spans over 10 km in altitude, it is important to stress that due to collisional quenching effects the fluorescence yield is also dependent on pressure, temperature and humidity of the air [10, 11]. More detail about recent relevant fluorescence yield measurements and a compilation of experimental results is available in a recent review [13].

The fluorescence detector (FD) comprises four observation sites — Los Leones, Los Morados, Loma Amarilla, and Coihueco — located atop small elevations on the perimeter of the SD array. Six independent telescopes, each with field of view of  $30^\circ \times 30^\circ$  in azimuth and elevation, are located in each FD site. The telescopes face towards the interior of the array so that the combination of the six telescopes provides  $180^\circ$  coverage in azimuth. Figure 3 shows the arrangement of the telescopes inside an observation site.

This arrangement of four FD sites was the optimum solution to the primary design goal of ensuring 100% FD triggering efficiency above  $10^{19}$  eV over the entire area of the surface detector. At the Malargüe site a large flat area, ideal for deployment of the surface detector, is bordered with a number of small hills suitable for FD sites. The arrangement of four inward-looking FD sites is a cost-effective way of ensuring full coverage without wasteful overlaps, and of minimizing the average distance to detected air showers, thus reducing uncertainties in atmospheric transmission corrections. Requiring “stereo” observations (showers triggering two FD sites) was not a strong design

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<sup>2</sup>This apparently unusual combination of two distinct experiments is determined by the fact that although the latter experiment, the AIRFLY, has already produced very detailed studies of relative band intensities, and of yield dependence on pressure, temperature and humidity, only the preliminary results for the absolute fluorescence yield were published [12].



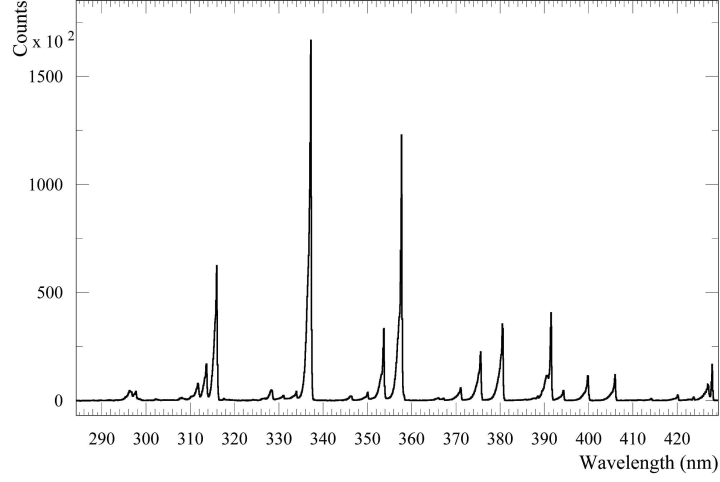


Figure 2: Measured fluorescence spectrum in dry air at 800 hPa and 293 K. From [10].

56 criterion because of the excellent “hybrid” geometry reconstruction available  
 57 when combining information from the FD and the SD (Section 7), although  
 58 stereo events are common at the highest energies.

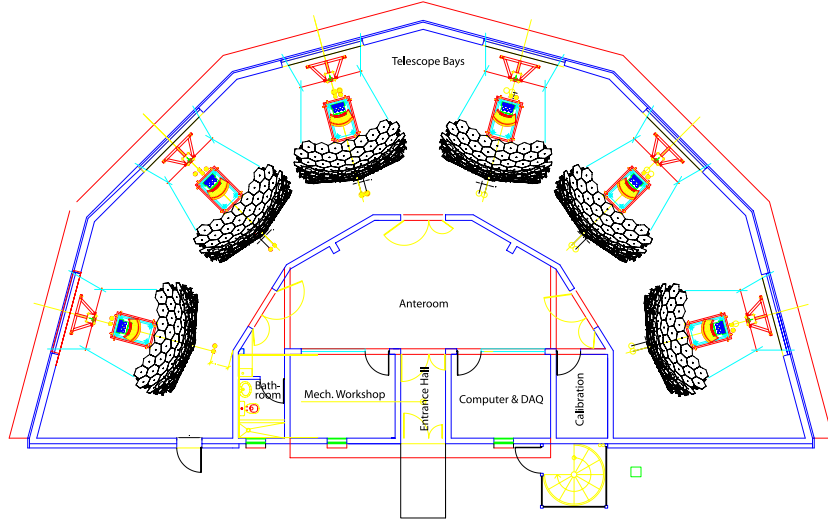


Figure 3: Schematic layout of the building with six fluorescence telescopes.

59 Figure 3 depicts an individual FD telescope. The telescope is housed in  
 60 a clean climate-controlled building. Nitrogen fluorescence light enters through

61 a large UV-passing filter window and a Schmidt optics corrector ring. The  
 62 light is focused by a 10 square meter mirror onto a camera of 440 pixels with  
 63 photomultiplier light sensors. Light pulses in the pixels are digitized every 100  
 64 nanoseconds, and a hierarchy of trigger levels culminates in the detection and  
 65 recording of cosmic ray air showers.

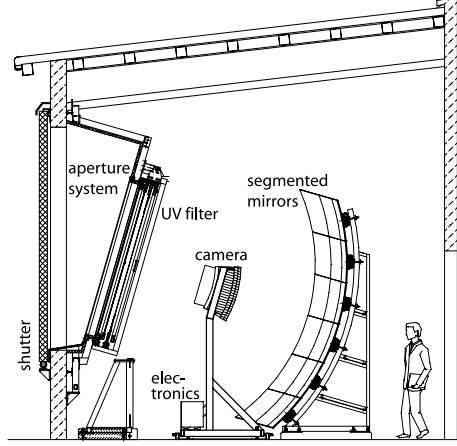


Figure 4: Schematic view of a fluorescence telescope of the Pierre Auger Observatory.

66 This paper is organized as follows. In section 2 we describe the components  
 67 of the optical system of an individual telescope, and in section 3 we focus on  
 68 the telescope camera. The electronics of a fluorescence telescope and the data  
 69 acquisition system (DAQ) of an FD station are described in section 4. The  
 70 details of the calibration hardware and methods are given in section 5, and  
 71 the performance, operation and monitoring of the fluorescence detector are  
 72 explained in section 6. Finally, in section 7 we concentrate on the basics of  
 73 shower reconstruction using the measured fluorescence signal, and in section 8  
 74 we summarize.

## 75 2. Optical system

76 The basic elements of the optical system in each FD telescope are a filter  
 77 at the entrance window, a circular aperture, a corrector ring, a mirror and a  
 78 camera with photomultipliers.<sup>3</sup> The geometrical layout of the components is  
 79 depicted in Fig. 6.

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<sup>3</sup>We remind the reader that the system is housed in a clean climate-controlled building, where an air-conditioning system is set to stabilize the temperature at 21°C. This helps to minimize thermal dilation of the system and to maintain the alignment of mechanical and optical components.

80 The window is an optical filter made of Schott MUG-6 glass [15]. This  
 81 absorbs visible light while transmitting UV photons from  $\sim 290$  nm up to  
 82  $\sim 410$  nm wavelength, which includes almost all of the nitrogen fluorescence  
 83 spectrum (see Figure 5). Without the filter window, the fluorescence signals  
 84 would be lost in the noise of visible photons.

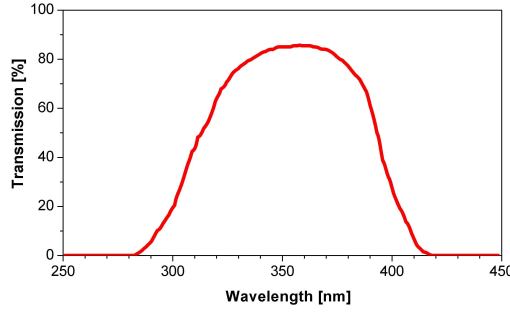


Figure 5: The measured transmission curve of the sample of the MUG-6 filter in the range between 250 nm and 450 nm.

85 The aperture, the corrector ring, the mirror, and the PMT camera constitute  
 86 a modified Schmidt camera design that partially corrects spherical aberration  
 87 and eliminates coma aberration. The size of the aperture is optimized to keep  
 88 the spot size<sup>4</sup> due to spherical aberration within a diameter of 15 mm, i.e. 90%  
 89 of the light from a distant point source located anywhere within the  $30^\circ \times 30^\circ$   
 90 FOV of a camera falls into a circle of this diameter. This corresponds to an  
 91 angular spread of  $0.5^\circ$ . In comparison, the FOV of a single camera pixel is  $1.5^\circ$ .  
 92 The light distribution within the spot is described by the point spread function  
 93 (PSF) shown in Fig. 7.

94 The schematic view of the spot size diagrams over the whole FOV is shown  
 95 in Fig. 8, where the rows correspond to viewing angles  $0^\circ$ ,  $10^\circ$ ,  $15^\circ$  and  $20^\circ$ .  
 96 The columns corresponds to different displacements of the camera off the focal  
 97 plane, by changing the camera-mirror distance from  $-5$  mm to  $+5$  mm with  
 98 respect to the nominal separation. The central position is located at a distance  
 99 of 1657 mm from the primary mirror. The asymmetric shape of some spots is  
 100 due to vignetting and camera shadow. This picture also shows the sensitivity

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<sup>4</sup>The image of the point source at infinity on the focal surface of the optical system is commonly called the “spot” in optics, but it may be better known as a “point spread function”. The size of the spot characterizes the quality of the optical system.

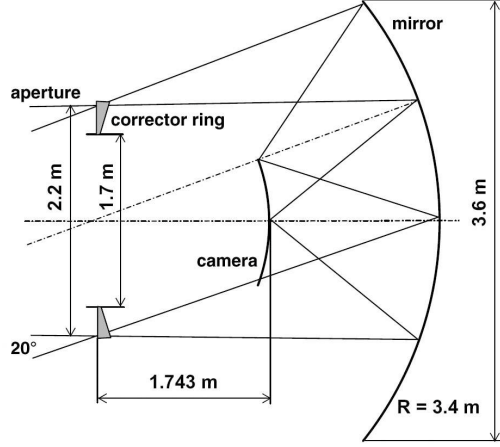


Figure 6: Geometrical parameters of the FD telescopes.

of the telescope PSF to the precision of the adjustment of the distance between mirror and camera.

### 2.1. Segmented mirror

Due to the large area of the primary mirror ( $\sim 13 \text{ m}^2$ ), the mirror is segmented to reduce the cost and weight of the optical system. Two segmentation configurations are used<sup>5</sup>: first, a tessellation of 36 rectangular anodized aluminum mirrors of three different sizes; and second, a structure of 60 hexagonal glass mirrors (of four shapes and sizes) with vacuum-deposited reflective coatings. In both cases, the spherical inner radius of mirror segments was specified as 3400 mm, and mirror segments with measured radii in the range between 3400 mm and 3420 mm were accepted and used for the construction.

The 12 telescopes at the Los Leones and Los Morados sites use aluminum mirrors. The mirror elements were produced from 20 mm-thick cast aluminum blocks, and were milled to the required spherical inner radius using a diamond milling technique. After the initial milling, the reflective surface was created by gluing a 2 mm sheet of AlMgSiO<sub>5</sub> alloy onto the inner surface of each element. The sheets were attached to the aluminum block at elevated temperature and pressure to achieve high stability. The material used is well-suited for high-precision diamond milling, and allows the final mirror surface to achieve a roughness below 10 nm. Finally, a 90 nm thick aluminum-oxide layer was applied to the surface by chemical anodization to provide additional protection.

<sup>5</sup>The two different configurations were used due to production reasons, the production was divided into two laboratories, where one laboratory was specialized in the production of aluminum mirrors and the other one in the production of glass mirrors. Thus, the production was faster and more cost-effective.

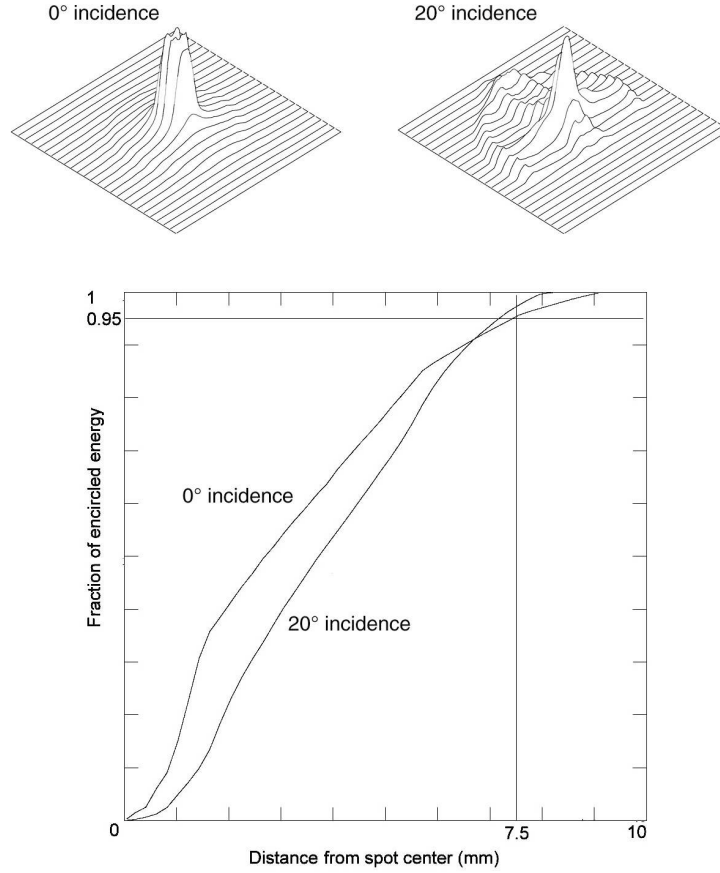


Figure 7: *Top*: The simulated 3D distribution of the light intensity for spots on the optical axis (left) and close to the camera corner (right; distance from the optical axis =  $20^\circ$ ). The size of the imaged area is  $20 \times 20$  mm. *Bottom*: Fraction of encircled energy as a function of spot diameter for the spot on the optical axis (upper curve; angular distance from the optical axis =  $0^\circ$ ; this curve corresponds to the distribution shown top left) and for the spot close to the camera corner (lower curve; angular distance from the optical axis =  $20^\circ$ ; this curve corresponds to the distribution shown top right).

122 The remaining 12 telescopes in the buildings at Coihueco and Loma Amarilla  
 123 use mirrors with glass segments. The thickness of the glass is 12 mm, and it  
 124 is composed of SIMAX, a borosilicate glass of the PYREX type. The choice  
 125 of lightweight glass was motivated by the need to maintain the optical stability  
 126 of the segments at different temperatures and for different inclinations of the  
 127 individual segments. SIMAX is suitable for machine grinding and polishing,  
 128 and has good mechanical and thermal stability.

129 The reflective layer on the glass mirror segments is composed of Al layer, with

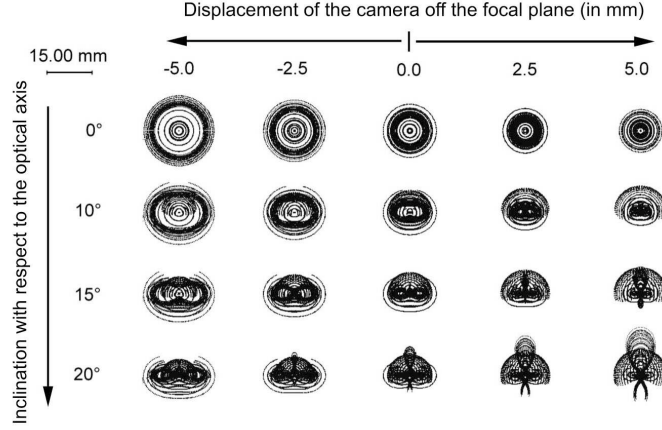


Figure 8: Spot diagrams of the telescope.

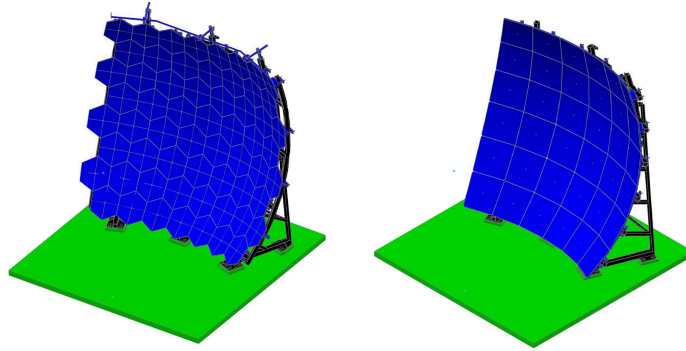


Figure 9: Two different segmentation configurations of the telescope mirror: *left*: 60 hexagonal glass mirror segments; *right*: 36 rectangular anodized aluminum mirror segments.

130 thickness of 90 nm, covered by one  $\text{SiO}_2$  layer with thickness of 110 nm. The  
 131 aluminum is used for its high reflectivity, and the silica for its high mechanical  
 132 resistance. The reflectivity has been measured at several points of each segment,  
 133 and the average reflectivity at  $\lambda = 370$  nm exceeds 90%. The surface roughness  
 134 has been measured mechanically and optically, and has an RMS of less than  
 135 10 nm.

136 In the glass mirrors, the mirror segments are grouped by their curvature  
 137 radii to minimize the radius deviations in each telescope. The curvature and  
 138 the shape of the reflective surface of the segments are measured and controlled  
 139 using standard Ronchi and/or Hartmann tests [16].

140 The alignment of the mirror segments is accomplished by directing a laser

141 onto each segment. The laser is mounted on the center of curvature of the  
 142 full mirror. The laser beam approximates a point source, and each segment  
 143 is adjusted such that the beam is reflected to a common point (the center of  
 144 the laser aperture). This procedure is used to adjust all the mirror segments  
 145 and match the full mirror to the required spherical shape, with the center of  
 146 curvature aligned with the optical axis of the telescope.

## 147 *2.2. Corrector ring*

148 A novel solution of the optical system with a “corrector ring” was designed  
 149 to keep the advantage of a large aperture of the Schmidt system, and  
 150 simultaneously simplify the production of the element, minimize its weight and  
 151 cost, while maintaining the spot size within the limits of aforementioned design  
 152 specification [17]. The aperture area of the telescope with the corrector ring  
 153 is almost doubled with respect to the optical system without any correcting  
 154 element. The analysis of real shower data [18] has compared the performances  
 155 of FD optical systems with and without corrector ring<sup>6</sup>, and has verified that  
 156 the corrector ring enhances the FD aperture by a factor of  $\sim 2$ .

157 The corrector ring is the circumferential part of the corrector plate of a  
 158 classical Schmidt camera with one planar side and the other with an aspheric  
 159 shape corresponding to a 6<sup>th</sup>-order polynomial curve. Such a surface is difficult  
 160 to manufacture and therefore some optimizations were adopted to simplify  
 161 the lens production. Eventually, a spherical approximation specially designed  
 162 for the fluorescence detector was chosen to fulfill both price and performance  
 163 requirements (see Fig. 10). The simplified corrector ring is located at the  
 164 aperture and covers the annulus between radii of 0.85 m and 1.1 m.

165 Since the rings have an external diameter of 2.2 m, their manufacture and  
 166 transportation to the site in a single piece would have been very difficult.  
 167 Therefore, each lens was divided into 24 segments. The size and profile of  
 168 one segment is shown in Fig. 11. The production of the segments has been  
 169 performed by Schwantz Ltd [19] after assembling a machine with a circular  
 170 base to hold the segments, and a disk with diamond abrasive cylinders for the  
 171 grinding of the glass (BK7 glass from Schott [15]) to the desired profile. After  
 172 production of each ring, its segments were tested for proper shape. To scan the  
 173 ring profiles, a laser beam was pointed towards the curved and flat surfaces and  
 174 the positions of the reflected light from each surface were measured [17].

## 175 *2.3. Simulation of the optical system*

176 To put our theoretical expectations about the performance of individual  
 177 optical components in context, a dedicated complete simulation of the optical  
 178 system was developed using Geant4 [20, 21], a Monte Carlo toolkit for the  
 179 simulation of radiation and light propagation. The tracking of optical photons  
 180 includes refraction and reflection at medium boundaries, Rayleigh scattering and

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<sup>6</sup>Test measurements without corrector ring were realized on several FD telescopes prior to corrector ring installation.



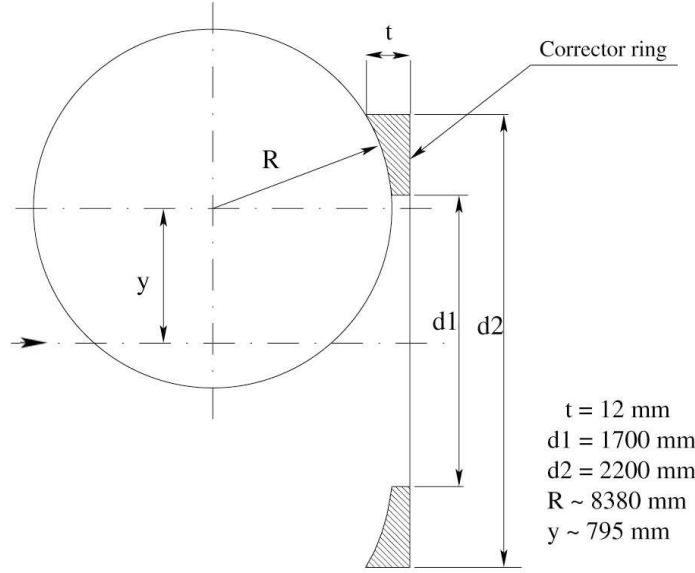


Figure 10: Corrector ring profile using a spherical approximation, the optical axis of the telescope is identified by the dash-dotted line with an arrow on the left and  $d2$  is the diameter of the aperture.

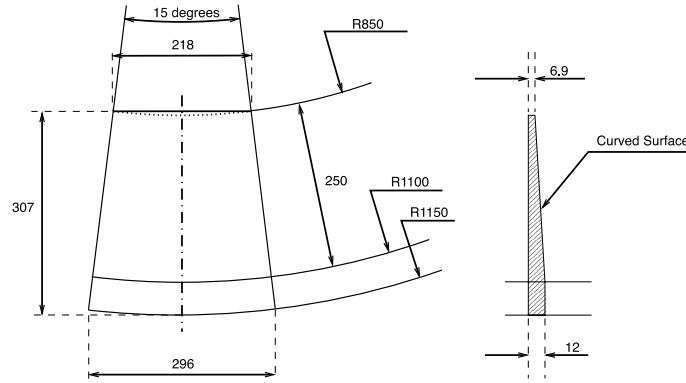


Figure 11: Technical drawing of one corrector ring segment. The dimensions are given in millimeters.

181 bulk absorption. The optical properties of a medium, such as refractive index,  
 182 absorption length, and reflectivity coefficients, can be expressed as functions of  
 183 the wavelength. The application includes a detailed description of the different

184 detector elements – UV filter, corrector lens, mirror and camera. The optical  
 185 properties of all materials, such as the absorption length and the refractive  
 186 index, were implemented as a function of wavelength.

187 The results of simulation are in agreement with all design specifications  
 188 of the FD optical system, i.e. even in the corners of the camera 90% of  
 189 the light from a distant point source is concentrated within a diameter of 15  
 190 mm. Moreover, the power of Geant4 simulation was recently proven during the  
 191 ongoing experimental study of the point spread function using distant flasher.  
 192 The Geant4 simulation has correctly predicted even the faint features of the  
 193 scattered light pattern, e.g. the ghost spot due to the reflection on the corrector  
 194 ring and filter.

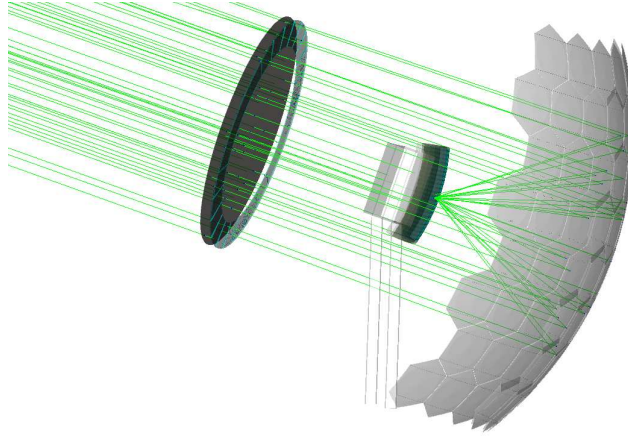


Figure 12: Ray tracing simulation of the optical system of the telescope made using Geant4.

### 195 3. Telescope camera

196 The camera [22] is the sensitive element of a telescope. It is composed of a  
 197 matrix of 440 pixels located on the focal surface of the telescope. The camera  
 198 pixels are arranged in a matrix of 22 rows by 20 columns (Fig. 13c). The  
 199 corresponding field of view is of  $30^\circ$  in azimuth (full acceptance of one row) and  
 200  $28.1^\circ$  in elevation (full acceptance of one column).

201 A cosmic ray shower is imaged on the camera as a line of activated pixels  
 202 having a track-like geometrical pattern and also a clear time sequence. Each  
 203 pixel is realized by a photomultiplier with a light collector.

#### 204 3.1. Geometry

205 The pixel array lies on the focal surface of the optical system, which is a  
 206 sphere of 1.743 m radius. The pixels are hexagonal with a side to side distance of  
 207 45.6 mm, corresponding to an angular size of  $1.5^\circ$ . The pixel centers are placed

208 on the spherical surface following the procedure outlined in Fig. 13, where we  
 209 define  $\Delta\theta = 1.5^\circ$  and  $\Delta\phi = 0.866^\circ$ . The first center is placed at  $+\Delta\theta/2$  with  
 210 respect to the telescope axis, which is taken as the  $z$  axis in Fig. 13. The other  
 211 pixel centers are obtained with increasing (or decreasing)  $\Delta\theta$  steps. In this  
 212 manner, a row of twenty pixels (corresponding to  $30^\circ$  in azimuth) is built.  
 213 The following row of pixels is obtained by a rotation of  $\Delta\phi$  around the  $x$  axis.  
 214 To produce the correct staggering between rows, the pixel centers are moved by  
 215  $\Delta\theta/2$  with respect to their positions in the previous row.

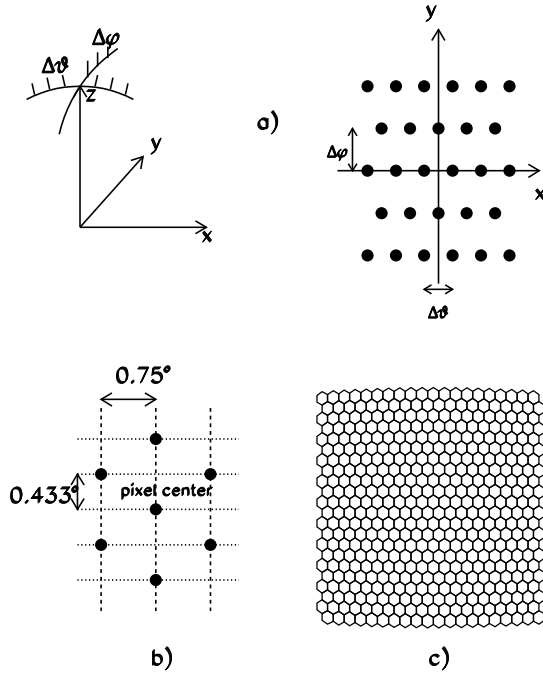


Figure 13: Geometrical construction of the FD camera; a) pixel centers are placed on the  
 spherical surface in steps of  $\Delta\theta = 1.5^\circ$  and  $\Delta\phi = 0.866^\circ$ , b) positioning of the pixel vertices  
 around the pixel center, c) the camera with 440 pixels arranged in a 22x20 matrix.

216 Once the pixel centers have been defined, each pixel hexagon is determined  
 217 by positioning six vertices around the center. The angular positions of the  
 218 vertices are obtained by moving in steps of  $\Delta\theta/2$  and  $\Delta\phi/3$  with respect to the  
 219 pixel center, as depicted in Fig. 13b. Equal steps in angle produce different linear  
 220 dimensions depending on the pixel position on the spherical surface. Therefore,  
 221 the pixels are not exactly regular hexagons, but their effective area varies over  
 222 the focal surface by 3.5% at most. Differences in the side length are smaller  
 223 than 1 mm, and are taken into account in the design of the light collectors (see  
 224 section 3.3) and in the analysis.

### 3.2. Mechanics

The camera body was produced from a single aluminum block by a programmable milling machine. It consists of a plate of 60 mm uniform thickness and approximately rectangular shape (930 mm horizontal  $\times$  860 mm vertical), with spherical outer and inner surfaces. The outer radius of curvature is 1701 mm, while the inner radius is 1641 mm. Photomultiplier tubes are positioned inside 40 mm diameter holes drilled through the plate on the locations of the pixel centers. Small holes in the camera body at the pixel vertices are used to secure the light collectors in place. A picture of the camera body is shown in Fig. 14.

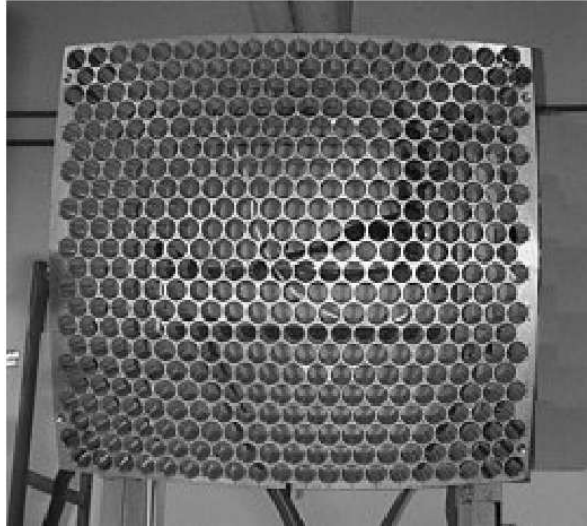


Figure 14: Picture of the camera body.

Within a finished telescope, the camera body is held in place by a simple and robust two-leg steel support made of 5 cm wide C-shaped steel beams. The obscuration of the mirror FOV due to the camera support is less than one-tenth of that of the camera. Power and signal cables run inside the C-shaped legs of the support without producing additional obscuration.

#### 3.2.1. Mechanical precision and alignment

To align an FD telescope, the pixel surface must be placed at the correct longitudinal distance from the center of curvature of the mirror, while the camera body should be centered on and perpendicular to the telescope axis, with the top and bottom sides of the camera parallel to the ground. Ray tracing computations have shown that the spot size increases by about 10 % when the longitudinal distance from the center of curvature changes by  $\pm 2$  mm. Therefore, the accuracy on the longitudinal position of the pixels on the focal surface should be better than  $\pm 2$  mm.

The intrinsic accuracy of the rigid metal frame of the camera body is very good, at the level of  $\pm 0.1$  mm. The point-to-point internal accuracy of the pixels on the camera body is at the level of  $\pm 1$  mm in both the longitudinal and transverse directions, due to the positioning of the photomultipliers and of the light concentrators on the rigid frame of the camera body.

The fluorescence buildings are surveyed by standard topographic methods and for each bay two accurately determined reference points are marked on the floor defining a line which corresponds to the azimuth of the telescope axis. The center of the camera is placed in the vertical plane containing this line and at the right nominal height. The camera is then aligned horizontally using a digital level-inclinometer with a precision better than  $0.05^\circ$  which is placed on the top of the camera body. Once the center of the camera is correctly positioned and the camera is horizontal, the right orientation is achieved measuring the distance of the four corners of the camera body to the mechanical reference point which is located at the mirror center. These measurements are done with a commercial laser distance-meter with a precision of  $\pm 1$  mm mounted on the mirror center. The alignment procedure provides a positioning with space accuracy at the level of  $\pm 1$  mm and angular precision at the level of one millirad i.e. better than  $0.1^\circ$ .

Measurements of the image of bright stars on the camera focal surface have verified that the alignment procedure for mirror elements and camera body meet the design specifications [23]. The precision of the absolute pointing of the telescopes has been checked to within an accuracy of  $0.1^\circ$ . The alignment is also being monitored using the dedicated inclined shots from the CLF. The more detailed studies are ongoing, however up to now both methods have detected no time variations of the telescope alignment. Also the stability of the reconstruction of the hybrid and stereo cosmic-ray events confirms that the camera and optical system of the telescope are stable within design specifications.

### 3.3. The pixel array

The hexagonal photomultiplier tube (PMT), model XP3062 manufactured by Photonis [24] is used to instrument the camera. Although their hexagonal shape represents the best approximation to the pixel geometry, a significant amount of insensitive area is nevertheless present between the photocathodes. In fact, some space between the PMTs is needed for safe mechanical packaging on the focal surface; moreover, the effective cathode area is smaller than the area delimited by the PMT glass envelope. To maximize light collection and guarantee a sharp transition between adjacent pixels, the hexagonal PMTs are complemented by light collectors.

The basic element for the pixel light collector is a specially designed analog of a “Winston cone” realized by a combination of six “Mercedes stars”. Each Mercedes star has three arms oriented  $120^\circ$  apart, which is positioned on each pixel vertex. Six Mercedes stars collect the light for a given pixel. The geometrical structure of the light collector for one pixel is shown in Fig.15a.

The Mercedes stars are made of plastic material covered by aluminized Mylar<sup>TM</sup> foils. The arm length is approximately half of the pixel side length.

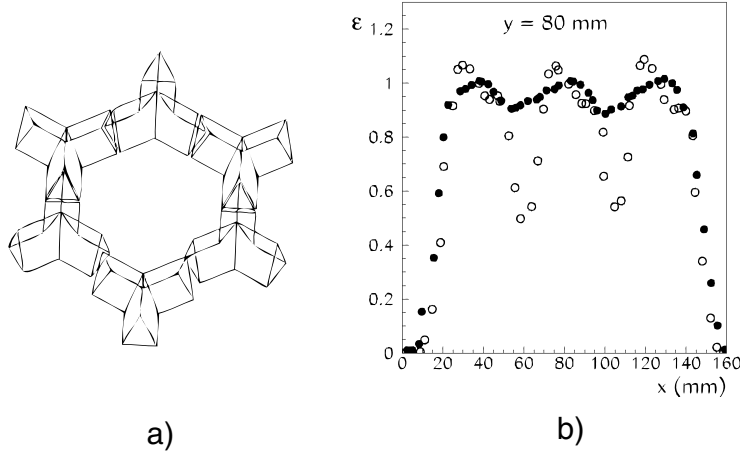


Figure 15: a) Six Mercedes stars positioned in order to form a pixel. Each Mercedes star has three arms oriented  $120^\circ$  apart. In the drawing the bottom Mercedes is slightly displaced for clarity. b) Measurement of the light collection efficiency, with a light spot moved along a line passing over three adjacent pixels. The full (open) dots represent the measurements performed with (without) Mercedes stars.

The arm section is an equilateral triangle. The base length of 9.2 mm is designed to cover the insensitive space due to photocathode inefficiency ( $\approx 2$  mm for each adjacent PMT) plus the maximum space between the glass sides of the PMTs (of the order of 5 mm). The triangle height is 18 mm, and the corresponding angle at the vertex is  $14.3^\circ$ .

Each Mercedes star is held by a bar, about 10 cm long, which is inserted into a 3.2 mm hole on the rigid frame of the camera body. The hole is located on the position of the pixel vertices (see Fig. 16a). On one end the bar is connected to the Mercedes star, while near the other end it is provided with a small O-ring. The bar is inserted in the hole from the camera front, and the O-ring comes out from the camera body, when the bar is fully inserted, keeping the Mercedes star in place. The geometry of the light collectors was designed on the basis of the optical system properties. The range of angles of incidence for the rays on the camera is in the interval between approximately  $10^\circ$  to  $35^\circ$ , as determined by the shadow of the camera and the aperture of the diaphragm. Note that the pixels are defined on the focal surface at 1743 mm from the center of curvature where the top edges of the Mercedes divide one pixel from its neighbors. The PMT cathodes are recessed behind that focal surface by 18 mm.

Monte Carlo ray tracing has shown that the light collection efficiency, averaged over the camera, is 94%, using for the reflectivity of the aluminized Mylar foils of the light collector surface the value of 85% measured at 337 nm. Without the light collectors, the collection efficiency decreases to 70%.

The Monte Carlo simulation was experimentally checked using a specially designed light diffusing cylinder with the exit hole for light rays. This device was

318 placed to a suitable fixed distance from the tested PMTs to acquire the similar  
 319 size of the spot as produced by the telescope optics. Also the distribution of the  
 320 angles of the light rays from this optical device is similar to that produced by  
 321 the telescopes. Measurements were made with and without the light collectors.  
 322 Results of a scan moving the light spot over the camera are shown in Fig. 15b.  
 323 Without light collectors, a significant loss of light at the pixel borders was  
 324 observed. When installed, the light collectors efficiently recuperate the light  
 325 loss. From these measurements, the light collection efficiency averaged over the  
 326 camera focal surface was found to be 93%.

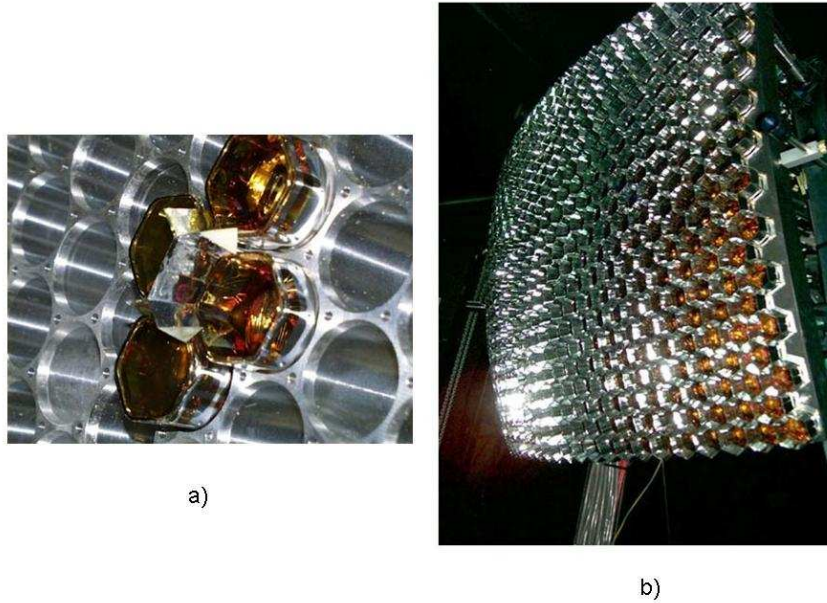


Figure 16: a) Detail of the camera body with four PMTs mounted together with two Mercedes  
 stars. The large holes to insert the PMTs and the small ones to mount the Mercedes are visible.  
 b) Picture of a camera completely assembled with all PMTs and light collectors in place.

327 A picture of a small section of the camera and a full picture of the camera  
 328 completely assembled are shown in Fig. 16.

### 329 3.4. The photomultiplier unit

330 The XP3062 photomultiplier [24] is an 8-stage PMT with a hexagonal  
 331 window (40 mm side to side). It is manufactured with a standard bialkaline  
 332 photocathode and its dependance of the quantum efficiency on the wavelength  
 333 is shown in Figure 17. The nominal gain for standard operation of the FD is  
 334 set to  $5 \times 10^4$ .

335 The PMT high voltage (HV) is provided by a HV-divider chain which forms  
 336 a single physical unit together with the signal driver circuitry. This unit,



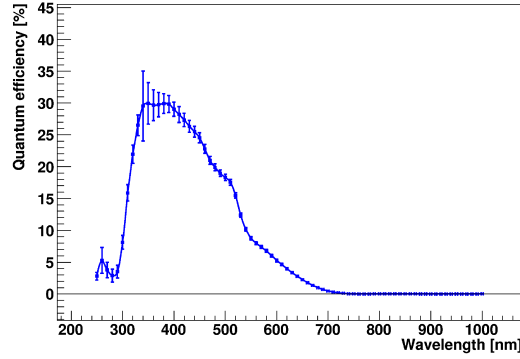


Figure 17: The measured dependence of the quantum efficiency on the wavelength of the Photonis XP 3062 photomultiplier unit.

337 called head electronics (HE), consists of three coaxially interconnected printed  
 338 circuits boards (PCBs): the bias PCB (innermost one), a laser-trimmed hybrid  
 339 technology driver circuit (intermediate), and the interface PCB (outermost).  
 340 The innermost and outermost PCBs are two-sided and of circular shape (32 mm  
 341 diameter), and are interconnected using high reliability pin connectors. The  
 342 HE units were manufactured by Intratec-Elbau (Berlin, Germany) [27] and are  
 343 soldered to the flying leads of the PMT. To ensure central mounting of the HE  
 344 with respect to the symmetry axis of the PMT, and to improve mechanical  
 345 rigidity, a specially-designed plastic structure has been introduced between the  
 346 glass tube and a central guidance hole left in the innermost PCB of the HE. A  
 347 PMT with the attached head electronics is shown in Fig. 18.

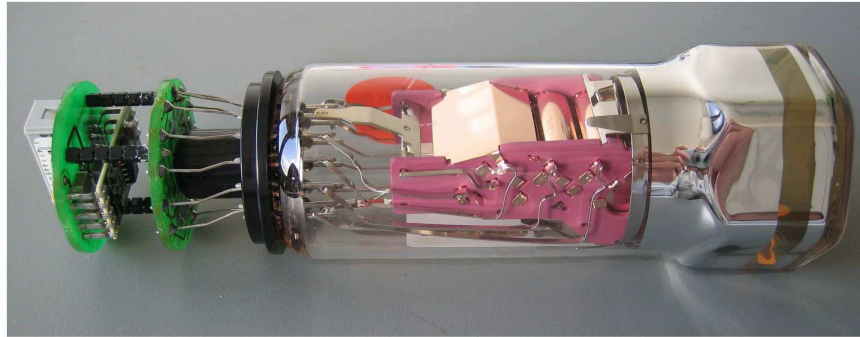


Figure 18: Picture of a PMT unit.

348 The HV divider keeps the PMT photocathode at ground and supplies  
 349 positive HV to the anode. To dissipate minimum power in the dense package of

HE units, the bleeder is operated at currents of less than  $170\,\mu\text{A}$ . Stabilization of the HV potential for large pulses and in presence of a relatively strong light background is thus realized by employing an active network [25] that uses bipolar transistors in the last three stages of the PMT. This is necessary for a fluorescence telescope, which is exposed to the dark sky background and in some cases also to a fraction of moon light. The normal dark sky background in moonless nights induces an anode current of about  $0.8\,\mu\text{A}$  on each PMT. The active divider ensures that the gain shift due to the divider chain is less than 1% for anode currents up to about  $10\,\mu\text{A}$ .

The driver located on the intermediate ceramics board of the HE receives the AC-coupled anode signal through a differential input. The anode pulse flows through a load resistor and reaches one leg of the differential input of the line driver, while the other leg only picks up the common-mode noise. A common-mode rejection ratio of 28 dB is obtained between 1 – 100 kHz. The integrated laser-trimmed hybrid circuit supplies a balanced output matching the  $120\,\Omega$  characteristic impedance of the twisted pair cable connecting the HE to the front-end analog board of the readout electronics (see section 4.1.1).

Extensive qualification and acceptance tests [26] were performed on the full PMT unit, i.e. the PMT with the HE attached, using automatic test systems. Among the tests were measurements of the HV dependence of the gain, which are needed for a proper grouping of the PMTs with similar gain. The relative photocathode sensitivity at different wavelengths, the linearity of the PMT unit, and the photocathode uniformity were also measured. Details are given in [26]. Before and after installation, the relevant information for each individual PMT unit is collected into a database which tracks the PMT characteristics, including calibration measurements performed *in situ*.

To reduce the cost of the power supplies, the photomultipliers of each camera are organized into ten groups of 44 units. Each group has similar gain characteristics, and is powered by a single HV channel. The spread of the gains within a group is about  $\pm 10\%$ .

Cables are distributed to the PMT electronics through distribution boards positioned behind the camera and within its shadow, i.e., without causing additional obscuration. These boards serve groups of 44 PMTs of the camera, supplying HV and LV and receiving the differential signals from the drivers in twisted pair cables. A fuse for an overcurrent protection of each LV line is provided on the board. From the boards, round shielded cables carry the PMT signals on twisted pair wires to the front-end crate, located at the base of the camera support.

#### 4. Electronics and data acquisition system

The FD telescopes are used to record fluorescence signals of widely varying intensity atop a sizeable, and continuously changing, light background. This presents a significant challenge for the design of the electronics and data acquisition system (DAQ), which must provide a large dynamic range and strong background rejection, while accepting any physically plausible air shower. The

394 DAQ must also allow for the robust, low-cost, remote operation of the FD  
 395 telescopes. And finally, the absolute FD-SD timing offset must be sufficiently  
 396 accurate to enable reliable hybrid reconstruction.

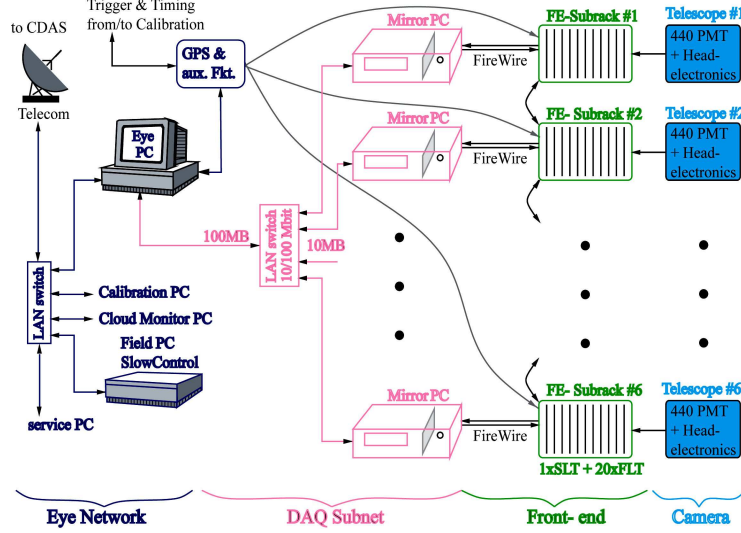


Figure 19: Readout scheme of an FD site The flow of recorded data is right to left.

397 The FD electronics are responsible for anti-alias filtering, digitizing, and  
 398 storing signals from the PMTs. As the PMT data are processed, they are  
 399 passed through a flexible three-stage trigger system implemented in firmware  
 400 and software. The remaining high-quality shower candidates are packaged by  
 401 an event builder for offline shower reconstruction. For each shower candidate  
 402 a trigger is sent online to the Observatory Central Data Acquisition System  
 403 (CDAS) in Malargüe. If at least one triggered surface detector station is found  
 404 in coincidence by CDAS, a hybrid trigger is generated and data of even a single  
 405 station is readout. An overview of the complete is presented in Table 1.

406 The organization of the electronics and DAQ is hierarchical, reflecting  
 407 the physical layout of the FD buildings. Figure 19 shows the readout  
 408 scheme of one FD site, divided into four logical units: head electronics for  
 409 440 PMTs  $\times$  6 telescopes, which provide low and high voltage; front-end (FE)  
 410 sub-racks, where the signals are shaped and digitized, and where threshold and  
 411 geometry triggers are generated; the DAQ subnet, in which six telescope PCs  
 412 (MirrorPCs) read out the stored data and perform additional software-based  
 413 background rejection; and an FD site network, in which a single computer  
 414 (named EyePC) merges triggers from the six telescopes and transfers them to  
 415 the CDAS in Malargüe. The FD site network also contains a Slow Control PC  
 416 to allow for remote operation of the building. The camera electronics and PCs  
 417 are synchronized by a clock module based on the Motorola Oncore UT+ GPS

418 receiver, the same receiver used in the SD array.

FD Trigger Sequence			
Trigger Level	Location	Purpose	Event Rate
1 (FLT)	FE sub-racks (FLT boards)	pixel threshold trigger	100 Hz pixel <sup>-1</sup>
2 (SLT)	FE sub-racks (SLT board)	track shape identification	0.1 – 10 Hz telescope <sup>-1</sup>
3 (TLT)	MirrorPCs (software)	lightning rejection	0.01 Hz telescope <sup>-1</sup>
T3	EyePC (software)	event builder, hybrid trigger	0.02 Hz building <sup>-1</sup>

Table 1: Trigger sequence for FD events. At each telescope, events are selected based on channel thresholds (FLT), track shape (SLT), and lightning rejection (TLT). Events passing the TLT are merged by an event builder on the FD EyePC. If the event passes further quality cuts for a simple reconstruction, a hybrid trigger (T3) is sent to CDAS.

#### 419 4.1. Front-end electronics

420 Each FD camera is read out by one front-end sub-rack and an associated  
 421 MirrorPC. The front-end electronics contain 20 Analog Boards (ABs), and each  
 422 AB receives data from a column of 22 PMT channels. The boards are designed  
 423 to handle the large dynamic range required for air fluorescence measurements;  
 424 at the energies of interest for the Pierre Auger Observatory, this means a range  
 425 of 15 bits and 100 ns timing [28]. The sub-racks also contain dedicated boards  
 426 for hardware triggers: 20 First Level Trigger (FLT) boards for pixel triggers,  
 427 and one Second Level Trigger (SLT) board for track identification within the  
 428 camera image.

##### 429 4.1.1. The Analog Board

430 The purpose of the AB is to receive inputs from the head electronics on the  
 431 PMT camera and adapt them for digitization by the analog-to-digital converters  
 432 (ADCs) located on a corresponding FLT board. As shown in Fig. 20, the analog  
 433 and FLT boards are physically connected by three 50-pin SMC connectors. The  
 434 combined front-end module measures 367 mm×220 mm, and is housed in a 9U  
 435 standard crate next to each FD telescope. Every crate contains 20 modules in  
 436 total.

437 One AB channel comprises the following logical blocks: receiver, gain stage,  
 438 anti-aliasing filter, and dynamic range adapter. The channel receives input from  
 439 the HE on the PMT camera via a distribution board. Individual pixel enabling  
 440 is performed by a fast analog switch, which is also used to generate an internal  
 441 test pulse upon request from the FLT logic. In each channel, the AB:

- 442 • performs a differential to single-ended conversion of the signal;

- adjusts the channel gain;
- applies an anti-aliasing filter before signal sampling;
- adapts the 15-bit dynamic range to the 12-bit ADCs;
- provides an injection point for test pulses.

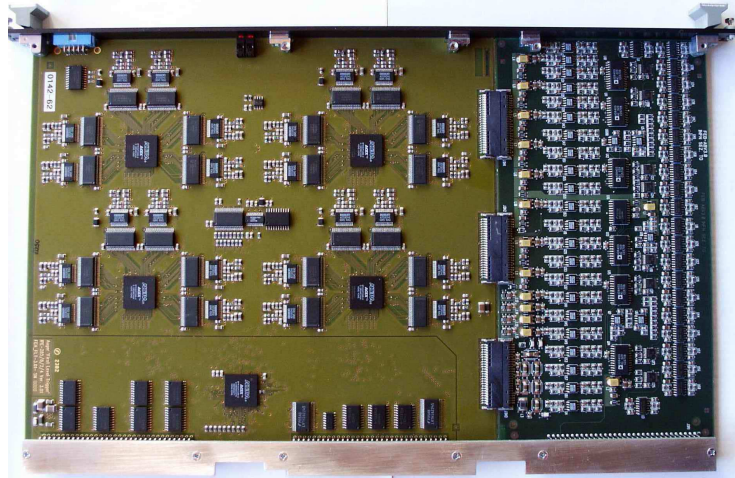


Figure 20: Photo of the FLT (left) and Analog Board (right): Both boards are connected by three 50-pin SMC connectors. A stiffener bar and common front panel provide mechanical rigidity.

The telescope PMTs are coarsely gain-matched during the installation of the PMT camera to avoid expensive individual HV control. Additional fine-tuning of the *channel* gains is achieved by means of digital potentiometers connected in series with a resistor on the feedback loop of each non-inverting gain stage. The devices can change individual channel gains up to a factor of about 1.9, and allow for gain matching of the channels in the camera to within 0.6%.

Prior to sampling, the PMT signal is processed by an anti-aliasing filter to match the 10 MHz digitization rate. A fourth-order Bessel filter with a cutoff frequency of 3.4 MHz has been implemented in the AB as a compromise between reconstruction error and circuit complexity. The Bessel filter scheme, featuring a linear dependence of the transfer function phase with frequency, was selected after a detailed study of optimal filters to minimize distortion of the current signal shape.

The final component of an AB channel is the dynamic range adapter. The FD records signal sizes between 3 and  $10^5$  photoelectrons per 100 ns, covering a dynamic range of 15 bits. Rather than digitize the full range of signal sizes, an optimal cost-effective solution using dynamic range compression to 12-bit ADCs has been implemented in the analog electronics design.

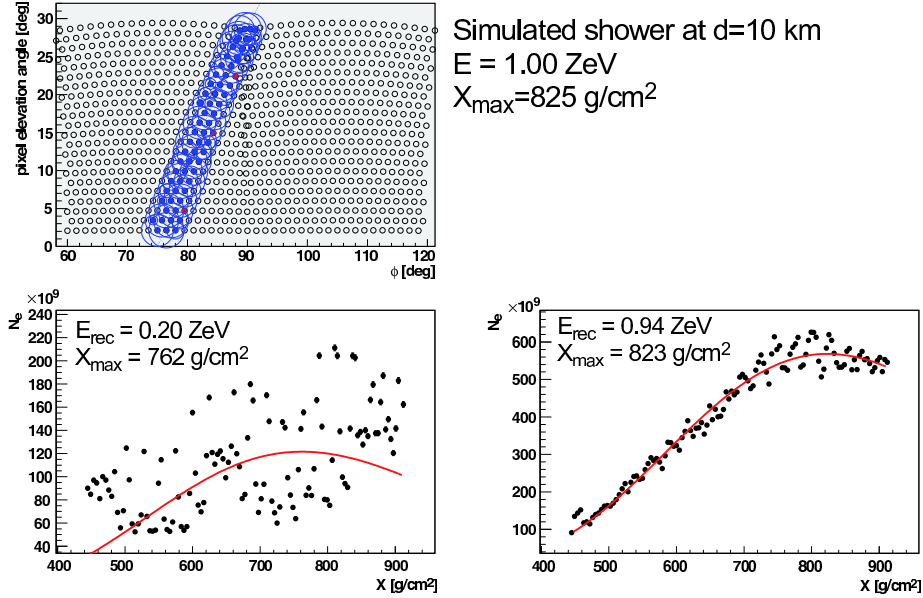


Figure 21: Simulated air shower of  $10^{21}$  eV at 10 km from the telescope. The top-left figure shows the light track in the cameras. The lower left figure shows the reconstructed shower profile without use of the virtual channel information. Making use of the virtual channel (lower right), the shower parameters can be reconstructed with minimal systematic distortions. The full lines in lower panels are fits of a Gaisser-Hillas function used for estimation of the shower energy (see Section 7.2).

The compression technique, which uses *virtual channels*, relies on the fact that the shower signal does not appear on all pixels simultaneously; instead, each pixel is triggered in a well-ordered time sequence. Every channel is configured with a high gain of about 20, and low gain of about 1. The high gain is optimized for the most frequent small- and intermediate-amplitude pulses, and is digitized pixel by pixel. The signal of the low gain stage is routed together from 11 non adjacent pixels in an analog summing stage. The sum signal is then processed by a virtual channel with a gain  $\sim 1$ .

When a nearby high-energy shower is detected by the FD, typically only one channel out of the group of 11 (which are routed together to the same analog stage) will saturate its 12-bit range at a given time. Even in the case of multiple saturation in the same group, such saturation will never occur in overlapping time bins. Therefore, the signal can be recovered from the virtual channel without ambiguity. This effectivity of this technique is demonstrated in Fig. 21, which depicts a simulated  $10^{21}$  eV shower with a landing point 10 km from the FD telescopes. The high-gain channels saturate during the development of the shower, and an attempt to reconstruct the shower using only these channels leads to significant systematic distortions in the shower profile. However, with the information in the virtual channel, the shower is reconstructed with almost



no systematic biases. More details about this technique and its results are available in [30].

#### 4.1.2. First Level Trigger (FLT) Module

The First Level Trigger (FLT) module is the heart of the digital front-end electronics. The module processes the data from one 22-channel column. Its main tasks are to:

- continuously digitize signals from the AB at 10 MHz;
- store the digitized raw data in memory for later readout;
- measure the pixel trigger rate for each channel;
- compensate for changing background conditions and maintain a pixel trigger rate of 100 Hz by adjusting the pixel trigger thresholds;
- allow access to raw data memory and internal registers;
- provide a digital interface to the AB to generate test pulses and to set the analog gain at the AB;
- calculate the baseline offset and its fluctuation averaged over a 6.5 ms period;
- calculate the multiplicity (number of triggered pixels) in one column.

The functions of the FLT are implemented in FPGA (Field Programmable Gate Array) firmware to improve the flexibility and cost-effectiveness of the module. A pipelined 12-bit ADC (ADS804) is used to digitize the signal at 10 MHz in each channel, and the data are stored with a 4-bit status word in 64k×16-bit SRAMs. The address space of each SRAM is divided into 64 pages of 1000 words. In the absence of triggers from the SLT module (described in the next section), each page works as a circular buffer to hold the ADC values of the previous 100  $\mu$ s. When an SLT trigger occurs, all FLT boards synchronously switch to the next unused memory page, whose address is provided by the SLT.

The main task of the FPGA logic is to generate the pixel trigger (FLT) using a threshold cut on the integrated ADC signal. The FLT is shown schematically in Fig. 22. A moving “boxcar” sum of the last  $n$  ADC samples is compared to a 14-bit threshold. Here  $n$  is a fixed number of time bins which can be chosen in the range  $5 \leq n \leq 16$ . The threshold is dynamically adjusted to maintain a pixel trigger rate of 100 Hz. When the sum exceeds the threshold, a pixel trigger is generated. The use of the sum substantially increases the signal-to-noise (S/N) ratio for each channel; for  $n = 16$ , S/N improves by a factor of 2.8 [31]. When the moving sum drops below the threshold, a retriggerable mono-flop extends the pixel trigger for an adjustable period of 5  $\mu$ s to 30  $\mu$ s common to all pixels, increasing the chance of coincident pixel triggers.

The rate of pixel triggers, called the “hit rate,” is measured in parallel by counters for each channel. It is used to adjust the threshold in such a



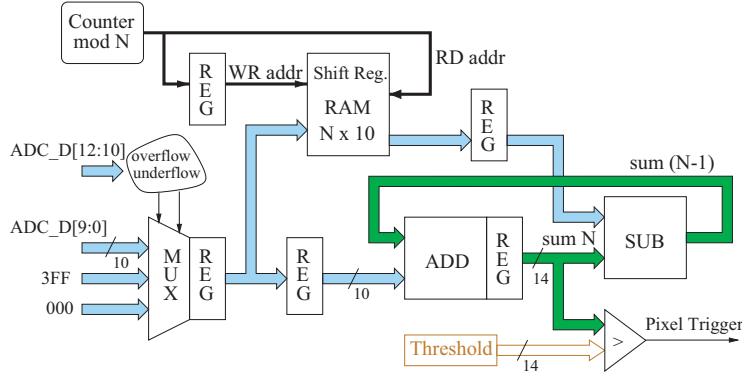


Figure 22: Generation of the First Level Trigger in one channel. A moving sum of 5 to 16 values is compared to an adjustable threshold. The use of the sum improves the S/N-ratio, and allows for regulation of the trigger rate.

manner that the hit rate is kept constant at 100 Hz under variable background light conditions. The background light levels seen by each PMT can also be monitored by analysis of the variances of the ADC values. Therefore, the FPGAs regularly evaluate the mean and variance of the channel baselines every 6.5 ms using 65536 consecutive ADC samples [33]. The measured channel hit rates, thresholds, offsets, and variances are periodically queried and stored in a monitoring database for experimental control (see section 6.5).

Finally, the multiplicity, or the number of pixels triggered simultaneously within 100 ns, is calculated for each FLT and for the full camera. The number of channels above threshold in one 22-pixel column is tracked on the FLT board. The total sum for the full camera is evaluated at the SLT board using daisy-chained lines on the backplane. The chronological sequence of multiplicity values carries information about the temporal development of the camera image, which is used by the software trigger (see section 4.2.1).

#### 4.1.3. The Second Level Trigger (SLT)

The pixel triggers generated for each channel in the 20 FLT boards of the FE sub-rack are read out by a Second Level Trigger (SLT) board. The functions of the SLT are implemented in FPGA logic, and its primary task is to generate an internal trigger if the pattern of triggered pixels follows a straight track [34]. The algorithm searches for track segments at least five pixels in length. It uses the fundamental patterns shown in Fig. 23, as well as those created by rotations and mirror reflections of these segments.

During data acquisition, some tracks will not pass through every pixel center, and therefore some PMTs along the track may not record enough light to trigger. To allow for this situation, and to be fault-tolerant against defective PMTs, the algorithm requires only four triggered pixels out of five. Counting all different four-fold patterns originating from the five-pixel track segments in Fig. 23, one

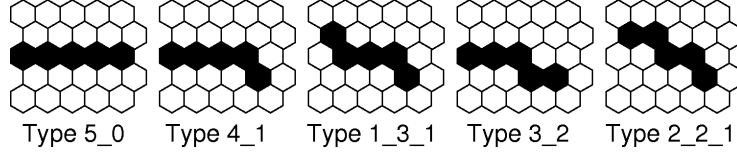


Figure 23: Fundamental types of pattern regarded as straight track segments.

550 finds 108 different pattern classes.

551 A brute-force check of the full  $22 \times 20$  camera pixel matrix for the 108 pattern  
 552 classes would result in 37163 combinations in total. Therefore, the SLT uses a  
 553 pipelined mechanism that scans for tracks on a smaller  $22 \times 5$  sub-matrix of the  
 554 camera. Every 100 ns, the FLT pixel triggers of two adjacent columns ( $22 \times 2$   
 555 pixels) are read into a pipelined memory structure. During the scan of the full  
 556 camera, coincidence logic identifies the pattern classes 1 to 108 and generates  
 557 a trigger if a pattern is found. The full scan lasts  $1 \mu s$ , and the pixel trigger  
 558 information scanned during this time is stored in the *pixel memory* of the SLT,  
 559 which is organized as a ring buffer for 64 events (as on the FLT). The FLT  
 560 multiplicities are also stored in the same memory, providing complementary  
 561 measures of the spatial and temporal size of the event.

562 Note that in addition to generating the geometrical trigger, the SLT board  
 563 also supplies bookkeeping information for the FE sub-rack. In particular, the  
 564 board provides a time stamp for each event using synchronization signals sent  
 565 by the GPS module over a dedicated line.

#### 566 4.1.4. Test Systems

567 The large number of channels in each telescope and the complexity of the  
 568 electronics and triggers require the use of an automatic testing system. The  
 569 testing and calibration of the full optics-PMTs-electronics chain is described in  
 570 section 5. However, the system also allows for tests of the electronics alone. A  
 571 test-pattern generator has been implemented on the front-end boards to create  
 572 pulses of variable width and amplitude at the input of each analog channel.  
 573 These test pulses are used to check the analog and digital functions of the FLT  
 574 and SLT boards. The system is independent of the full camera, and can be used  
 575 for maintenance when the HV/LV supply of the camera is turned off.

#### 576 4.2. DAQ Software and Software Trigger

577 Once an event has been processed by the DAQ hardware and stored in FLT  
 578 and SLT memory, it can be read out and analyzed by trigger software in the  
 579 MirrorPCs. Each MirrorPC is connected to the digital boards in the FE sub-  
 580 rack via a FireWire interface [29], and the MirrorPCs communicate with the  
 581 EyePC through a 10/100 Mbit LAN switch.

582 The DAQ system can handle data from three types of sources:

- 583 1. external triggers;
- 584 2. calibration events and test pulses;
- 585 3. real air shower events.

586 External triggers are primarily artificial light sources used for atmospheric  
 587 monitoring, such as laser shots from the Central Laser Facility [14]. The various  
 588 types of calibration pulses are described in section 5.

589 Events triggered externally, or by the calibration and test-pulser systems,  
 590 receive special trigger bits in the SLT status word to distinguish them from  
 591 real air shower data. When the MirrorPCs read out such events from the FE  
 592 modules, they are sent directly to the EyePC for event building and storage in  
 593 dedicated raw data files. In contrast, when the MirrorPCs read out true air  
 594 shower data, the events are processed by a Third Level Trigger. The surviving  
 595 data are sent to the EyePC, which builds an event from the coincident data in  
 596 all telescopes and generates a hybrid trigger (T3) for the surface array.

#### 597 4.2.1. The Third Level Trigger (TLT)

598 The Third Level Trigger (TLT) is a software algorithm designed to clean  
 599 the air shower data stream of noise events that survive the low-level hardware  
 600 triggers. It is optimized for the fast rejection of triggers caused by lightning,  
 601 triggers caused by muon impacts on the camera (see Fig. 34), and randomly  
 602 triggered pixels.

603 During the austral summer, distant lightning can significantly disrupt the  
 604 normal data acquisition of the FD telescopes. In good atmospheric conditions,  
 605 the SLT will detect one to two events per minute per telescope. However,  
 606 lightning can cause large parts of the camera (i.e., hundreds of pixels) to trigger  
 607 in bursts of several tens of events per second. If not filtered, the bursts will  
 608 congest the 64-event buffers in the FLT and SLT boards, and the telescope  
 609 readout (including ADC traces) will suffer a substantial increase in dead time.

610 The TLT is designed to efficiently filter lightning events without performing  
 611 costly readouts of the complete ADC traces. The algorithm achieves high speed  
 612 by reading out only the FLT multiplicities and the total number of triggered  
 613 pixels. Using several thousand true showers and background events from one  
 614 year of data, a dedicated study was performed to determine efficient lightning  
 615 cuts based on the time development of the multiplicity and the integral of the  
 616 multiplicity over the whole event consisting of 1000 subsequent multiplicity  
 617 values [35]. With these cuts, approximately 99% of all lightning events are  
 618 rejected by the TLT during a short (50  $\mu$ s) decision time.

619 The multiplicity-based lightning rejection effectively removes noise events  
 620 with more than 25 pixels. The remaining events with smaller number of  
 621 triggered pixels, which are contaminated by random pixel triggers and muon  
 622 impacts, are filtered in a second step. The correlation between the spatial  
 623 arrangement and peak signal times of triggered pixels is used to discard noisy  
 624 channels far off the light track [35]. Therefore, in this situation full ADC traces  
 625 must be used to determine the pixel trigger times. However, this readout is

done only for less than 25 pixels, and therefore it does not fill the FLT buffers and thus it does not appreciably increase the telescope dead time.

The TLT performance has been validated with simulated showers and data recorded during different weather and sky background conditions. Although the exact behavior of the algorithm depends on the actual conditions, approximately 94% of all background events are correctly identified by the TLT. The fraction of true showers rejected by the trigger is below 0.7%.

#### 4.2.2. Hybrid Trigger (T3)

Events passing the TLT in each telescope are sent to the EyePC through the DAQ subnet. An event builder running on the EyePC merges coincident events from adjacent telescopes. The EyePC also sends a hybrid trigger, called a T3, to the CDAS in Malargüe.

The T3 acts as an external trigger for the surface array. Its purpose is to record hybrid events at low energies (below  $3 \times 10^{18}$  eV) where the array is not fully efficient and would not often generate an independent trigger. Hybrid events at these energies occur within 20 km of the FD buildings and usually do not trigger more than one or two SD stations. However, as discussed in section 7, even limited SD data are sufficient for high-quality hybrid reconstruction.

The T3 algorithm is used to calculate a preliminary shower direction and ground impact time with a simple online reconstruction. Once these data arrive at the CDAS, a request is sent to the SD for signals recorded close to the calculated impact time. For each T3 trigger, the SD stations nearest the FD building (comprising approximately one-quarter of the array) are read out. The FD and SD data are merged offline for subsequent hybrid analysis.

## 5. Calibration

### 5.1. Introduction

The reconstruction of air shower longitudinal profiles and the ability to determine the total energy of a reconstructed shower depend on being able to convert ADC counts to a light flux at the telescope aperture for each channel that receives a portion of the signal from a shower. It is therefore necessary to have some method for evaluating the response of each pixel to a given flux of incident photons from the solid angle covered by that pixel, including the effects of aperture projection, optical filter transmittance, reflection at optical surfaces, mirror reflectivity, pixel light collection efficiency and area, cathode quantum efficiency, PMT gain, pre-amp and amplifier gains, and digital conversion. While this response could be assembled from independently measured quantities for each of these effects, an alternative method in which the cumulative effect is measured in a single end-to-end calibration is employed here.

The absolute calibration of the fluorescence detectors uses a calibrated 2.5 m diameter light source (known as the “drum”) at the telescope aperture, providing same flux of light to each pixel – see section 5.2. The known flux from the light source and the response of the acquisition system give the required

668 calibration for each pixel. Each telescope is drum calibrated ca. once per year<sup>7</sup>.  
 669 Calibration takes ca. one night per two telescopes, so ca. 3 days per building  
 670 with six fluorescence telescopes.

671 In the lab, light source uniformity is studied using CCD images and the  
 672 intensity is measured relative to NIST-calibrated photodiodes [39]. For details  
 673 see sections 5.3 and 5.4. Use of the drum for gain adjustment and calibration  
 674 provides a known, uniform response for each pixel in each camera of the FD  
 675 detector. The average response of the FD is approximately 5 photons/ADC  
 676 bin.

677 Three additional calibration tools are used as well:

- 678 • Before and after each night of data taking a relative calibration of the  
 679 PMTs is performed. This relative calibration is used to track both short  
 680 and long term changes in detector response.
- 681 • The relative FD response has been measured at wavelengths of 320, 337,  
 682 355, 380 and 405 nm, defining a spectral response curve which has been  
 683 normalized to the absolute calibration.
- 684 • An independent check of the calibration in some phototubes is performed  
 685 using vertical shots from a portable laser in the field.

686 The sections below describe the hardware and use of these calibration  
 687 systems.

## 688 5.2. Light source and drum

689 The technique [36] is based on a portable 2.5 m diameter, 1.4 m deep, drum-  
 690 shaped light source which mounts on the exterior of the FD apertures (see  
 691 Fig. 24). The source provides a pulsed photon flux of known intensity and  
 692 uniformity across the aperture, and simultaneously triggers all the pixels in the  
 693 camera. Combination of a diffusively reflective surface inside the drum<sup>8</sup> and  
 694 of a diffusively transmitting Teflon sheet at the front side of the instrument  
 695 provides that the front surface of the drum operates as a good Lambertian.  
 696 This means that the light emitted per unit solid angle from any small area  $A$   
 697 of the drum surface depends only on the angle  $\theta$  with respect to the normal  
 698 direction according to  $I(\theta) = I_0 A \cos(\theta)$ . Looking at the disk from angle  $\theta$  and  
 699 distance  $d$  with a fixed solid angle  $\omega$ , the viewed disk area is  $d\omega / \cos(\theta)$ . So the  
 700 intensity is  $I(\theta) = I_0 d\omega$ . The intensity is independent of viewing angle for a  
 701 perfect Lambertian. Fig. 25 shows that pixels of a CCD camera (each seeing a  
 702 small portion of the drum surface) measure the same intensity regardless of the  
 703 viewing angle to the drum.

704 To produce diffuse light inside the drum, illumination is provided by a pulsed  
 705 UV LED [37] ( $375 \pm 12$  nm), mounted against the face of a 2.5 cm diameter

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<sup>7</sup>Efficiency of the calibration process improves with time.

<sup>8</sup>For the details of the arrangement inside the drum please see the next paragraph.

706  $\times 2.5$  cm long Teflon<sup>TM</sup> cylinder. The Teflon cylinder is mounted in a 15 cm  
 707 diameter reflector cup, which is mounted flush to the center of the drum front  
 708 surface, illuminating the interior and the back surface of the drum. The LED  
 709 is inserted down the axis of the drum from the back through a pipe. A silicon  
 710 detector attached to the opposite end of the teflon cylinder monitors the light  
 711 output for each pulse of the LED.

712 The drum was constructed in sections, using laminations of honeycomb core  
 713 and aluminum sheet. The sides and back surfaces of the drum interior are lined  
 714 with Tyvek<sup>TM</sup>, a material diffusively reflective in the UV. The reflecting surfaces  
 715 of the cup are also lined with Tyvek. The front face of the drum is a 0.38 mm  
 716 thick Teflon sheet, which transmits light diffusively.

### 717 5.3. Calibration of the drum

718 The absolute calibration of the drum light source intensity is based on UV-  
 719 enhanced silicon photodetectors, calibrated at NIST to  $\pm 1.5\%$ . While the small  
 720 surface area and low response of these detectors preclude detection of the small  
 721 photon flux from the drum surface directly, the photodiode calibration can be  
 722 transferred to a more sensitive PMT/DAQ system.

723 To establish the absolute flux of photons emitted from the drum surface, a  
 724 reference PMT is placed on the drum axis in the optical laboratory<sup>9</sup>, 14 m from  
 725 the surface. The LED light source in the drum is pulsed for a series of 5  $\mu$ s

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<sup>9</sup>Where no telescope is present.

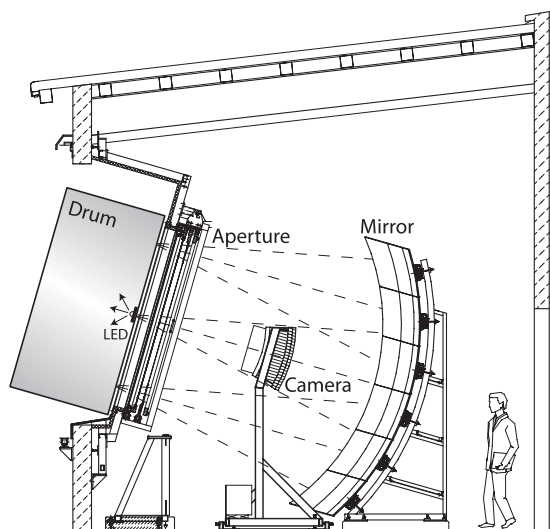


Figure 24: A schematic showing the drum mounted in a telescope aperture.

pulses and the charge from the PMT is integrated and recorded for each pulse, resulting in a histogram of the distribution of the observed integrated flux.

On an optical bench, the PMT is then exposed to a small diffuse LED light source with neutral density filter. This filter adjusts the light intensity to the level similar to that of the drum. The LEDs are pulsed with the same driving circuitry as for the drum, and the intensity is set to a series of values producing a series of histograms with PMT centroids surrounding that from the drum measurement above. At each of these intensity settings a second measurement is made in which the PMT is replaced by the NIST-calibrated photodiode and a neutral density filter in the source is removed, increasing the intensity to a level measurable by the photodiode. For this second measurement, the LEDs are run in DC mode. The relationship of PMT response to photodiode current is found to be very linear. The flux of photons at 14 m from the drum surface is calculable from the active area of the photodiode, the neutral density filter reduction factor, the LED pulsed/DC duty factor, the NIST calibration for the photodiode, and the value of photodiode current corresponding to the PMT-drum centroid. The value of photodiode current is interpolated from the linear response-current relationship given above. The resulting uncertainty in the drum intensity is 6%. Additional uncertainty contributions related to use of the drum at the FDs [41], such as temperature dependencies, along with camera response variations in time and spectral characteristics of the LED light source, combine with the drum intensity uncertainty resulting in an overall uncertainty of 9% for the absolute FD calibration.

#### 5.4. Drum relative uniformity measurements

Uniformity of light emission from the drum surface is important, since the pixels in a FD camera view the aperture at varying angles. In addition, for each pixel, a different part of the aperture is blocked by the camera itself. Studies were made of uniformity of emission across the surface and as a function of viewing angle. These uniformity measurements were made using a CCD, viewing the emitting surface of the drum from a distance of 14 m (see Fig. 25). Images were recorded with the drum axis at angles of 0, 10, 20 and 25 degrees relative to the CCD axis, covering the range of the Auger telescope field of view ( $0 - 21^\circ$ ). For these images, the UV LEDs were powered continuously.

A  $0^\circ$  image was used to analyze the uniformity of emission over the drum surface. Using software, concentric circles were drawn on the image, defining annular regions of increasing radius, as shown in the figure. The intensity of the pixels in each region was analyzed to obtain the intensity as a function of radius. In the area defined by the 2.2 m aperture diameter, the relative uniformity of intensity is constant over the area to about  $\pm 2\%$  except for a central dark spot. The variation with viewing angle of a section of the drum image is also shown in Fig. 25.

The measured drum non-uniformities are small (diagonal stripes in Fig. 25, intensity decrease with radius, etc.), indicating that the FD pixels see similar intensities integrated over the drum surface. While perfect drum uniformity is desirable, the present non-uniformities are acceptably small and well mapped



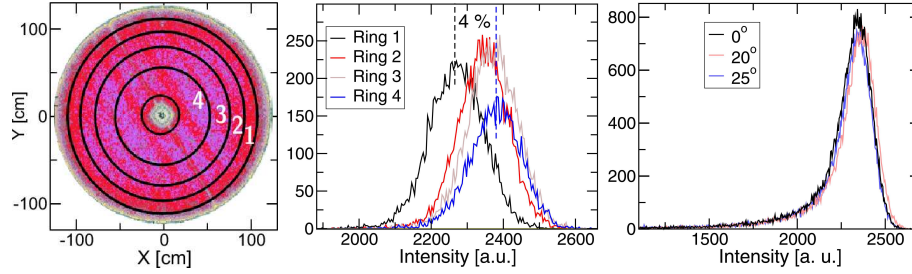


Figure 25: *Left*: CCD image at  $0^\circ$  drum angle, showing the defined rings for relative intensity analysis. Production deformations in the teflon material can be seen. *Middle*: A plot of the observed pixel intensities in the defined regions of the drum, shown in the left panel; *Right*: The results of angular relative intensity measurements at 0, 20, and  $25^\circ$  inclinations of the drum with respect to the CCD camera.

771 over the surface of the drum. A ray-tracing program using the uniformity and  
 772 angular intensity information from the CCD images shows less than 1% variation  
 773 in total flux seen by the pixels, and corrections are applied for these variations.

#### 774 5.5. Absolute calibration using vertical laser shots

775 The drum technique for absolute calibration has been checked for some pixels  
 776 using remote laser shots at 337 and 355 nm [40, 41]. A laser pulse is shot  
 777 vertically into the air with a known intensity. A calculable fraction of photons  
 778 is scattered to the aperture of the FD detector. This yields a known number of  
 779 photons arriving to the detector for each pixel, which views a segment of the laser  
 780 beam. The response of each pixel to the known number of photons constitutes an  
 781 absolute end-to-end calibration for those pixels. A calibrated laser probe is used  
 782 to measure the number of photons in typical laser pulses in the field calibration.  
 783 A nitrogen laser (337 nm) at a distance of 4 km is well suited for such purposes,  
 784 because the scattered light flux is in the correct range and uncertainties due to  
 785 aerosols are minimal. The scattering angles are just greater than  $90^\circ$ , where  
 786 the differential scattering cross section is minimized for aerosols. Hence the  
 787 Rayleigh-scattered light component is dominant. Moreover, at a distance of  
 788 4 km, the possible extra scattering from the beam by aerosols with respect  
 789 to scattering in aerosol free air is approximately canceled by the extra aerosol  
 790 attenuation of the primarily Rayleigh-scattered light flux. Therefore, the roving  
 791 laser provides an independent and redundant absolute calibration of the FD  
 792 cameras, based on the measured light injected into the atmosphere. Using the  
 793 recorded laser-induced ADC traces and the drum calibration constants, we can  
 794 then compare the number of predicted to measured photons.

795 Schematics of the hardware and controlling electronics for the roving laser  
 796 system are shown in Fig. 26. The roving laser system presently in use is based  
 797 on a 337 nm nitrogen laser, providing about  $100 \mu\text{J}$  per pulse, while the output  
 798 is inherently unpolarized. The RMS values of 100 shots in a set, fired over 200  
 799 seconds, is typically 3%. Variations of average energy in a 100 shot group is

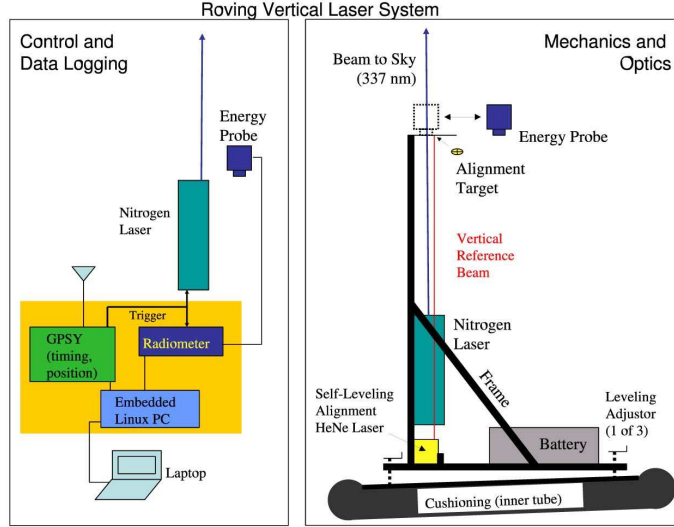


Figure 26: Roving vertical nitrogen laser hardware and control system.

2% on the time scale of 15 minutes, and 10% from night to night, as measured with the energy probe inserted into the beam for 100 shots before and after the calibration shots. A self-leveling He-Ne laser provides a reference beam within  $0.03^\circ$  of vertical, allowing alignment of the nitrogen laser to within  $0.1^\circ$  of vertical.

An embedded Linux PC, communicating via serial connection to a programmable GPS timing module, controls the laser firing time relative to the GPS second. The time stamp is used to identify calibration events for reconstruction.

For comparison with the 375 nm drum calibration, the measured relative FD response at the two calibration wavelengths must be known. The relative response has been measured, and is described in the next section.

Overall uncertainties of the roving laser calibration have been assessed at 12%<sup>10</sup>.

### 5.6. Multi-wavelength calibration

For calibration at wavelengths spanning the FD acceptance, a xenon flasher is mounted at the back of the drum, with a filter wheel containing 5 notch filters for selection of wavelengths. The xenon flasher [42] provides 0.4 mJ optical output per pulse covering a broad UV spectrum, in a time period of a few hundred nanoseconds. A focusing lens at the filter wheel output maximizes

<sup>10</sup>The total uncertainty of 12% has the following sources added in quadrature: 10% calibration of the laser energy, 7% aerosol contamination, Rayleigh calculation 1% and position uncertainty 2%.

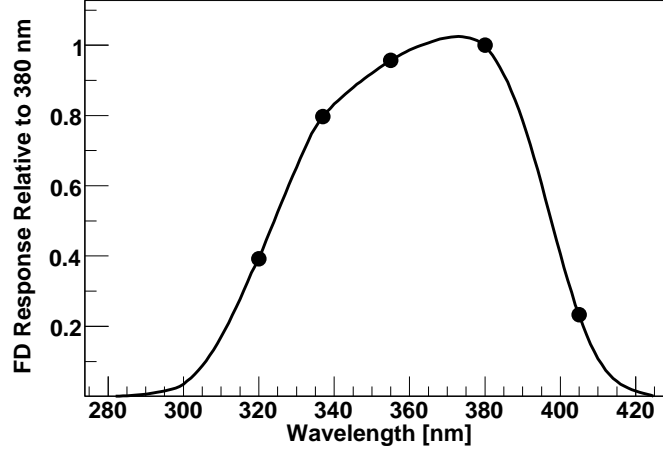


Figure 27: The results of the multi-wavelength measurements (see text, and [44].)

the intensity through the filter wheel and into the light pipe which is installed between Teflon diffusor and back of the drum.

Relative drum intensity measurements at wavelengths of 320, 337, 355, 380 and 405 nm have been made with the same reference PMT used in the absolute measurements. At each wavelength the recorded response from the reference PMT, combined with the PMT quantum efficiency and corrected for light source and filter width effects, yields a quantity that is proportional to the number of photons emitted from the drum. The FD response detected using the various filters, with the drum placed in the aperture, can be combined with the results from the laboratory to form the curve of relative camera response [44] shown in figure 27.

The curve in figure 27 is the result of an iterated spline fit beginning with a response curve predicted from manufacturer's specifications for FD components. The shape of this initial curve is dominated by the FD PMT QE and the UV filter transmission. The final fit includes effects of the notch filter transmission widths (15 nm FWHM), the reference PMT QE, the xenon light source emission spectrum, and the relative drum intensity for each filter, all measured in the lab, and the observed FD response to the drum for each filter. The relative uncertainty at each wavelength on the curve is 5%.

#### 5.7. FD relative calibration system

The relative optical calibration system [41] is used to monitor detector response and to track absolute calibration between drum calibrations. The system is used before and after each night of operation. Three positions (A, B, and C) are illuminated for each camera, monitoring different groups of detector

843 components. Light is distributed through optical fibers, from permanently  
844 installed light sources. All components use quartz optics.

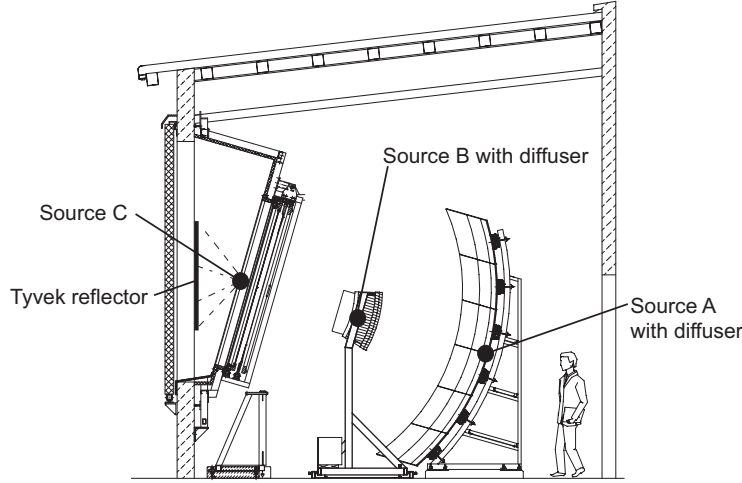


Figure 28: A schematic showing positions of light sources for three different relative calibrations of the telescope.

845 The A fiber light source is a 470 nm LED [38]. A 7:1 splitter at the source  
846 provides light to 6 fibers, running to a 1 mm thick Teflon diffuser located at  
847 the center of each mirror in the FD building, illuminating the camera face.  
848 The remaining fiber provides light for an output monitoring photodiode at the  
849 source. The A fiber LED is driven by a constant current source circuit. Normal  
850 operation pulses the LED for a series of 60  $\mu$ s pulses.

851 Each of the B and C light sources are xenon flash lamps [42]. Each source is  
852 mounted at the focus of a f/1.5 lens, with downstream optics including a beam  
853 splitter (for a source monitoring fiber), a filter wheel, and an f/2.4 lens focusing  
854 onto a 7:1 splitter. The 7 fibers run to each fluorescence telescope, and to a  
855 monitor for each output, as for the A fiber. The B source fibers are split near  
856 each camera and terminate at 1 mm thick Teflon diffusers located at the sides  
857 of the camera, with the light directed at the mirror. The C source fibers are  
858 also split, and terminate outside the aperture with the light directed outwards.  
859 Tyvek sheets are mounted on the inside of the aperture shutters. The sheets  
860 are positioned such that they are opposite the fiber ends when the shutters are  
861 closed, and the diffuse light scattered off the Tyvek enters the aperture.

862 The B source includes a Johnson-U filter [43], approximating the full  
863 wavelength acceptance of the fluorescence telescopes. The C source filter wheel  
864 containing interference filters is centered at wavelengths of 330, 350, 370, 390 and  
865 410 nm, for monitoring detector stability at wavelengths spanning the spectral  
866 acceptance.

## 867 **6. Performance, operation and monitoring of the detector**

868 All four fluorescence sites have been completed and are in operation. Los  
869 Leones has been in full operation since March 2004 and Coihueco since July  
870 2004. Los Morados began data acquisition in April 2005, and the fourth site at  
871 Loma Amarilla started its operation in February 2007.

### 872 *6.1. Uptime Fraction*

873 The operation of the fluorescence detector can be characterized by the  
874 uptime, or the fraction of the total time in which the FD was acquiring data.  
875 Anything that disables the measurement is considered as dead time. The main  
876 contributions to the dead time are the presence of the sun and nearly full moon  
877 on the sky, poor weather and the presence of the moon in any phase within  $5^\circ$  in  
878 the FOV of a telescope. The position of the moon can be calculated in advance  
879 and the shutters of individual telescopes are closed when the moon approaches.

880 The shutters are also automatically closed when the weather conditions  
881 become dangerous for operation (high wind speed, rain, snow, etc.) and when  
882 the observed sky brightness (caused mainly by scattered moonlight) is too high.  
883 The influence of weather effects depends on the season, with the worst conditions  
884 typically occurring during Argentinian summer. Other periods of dead time are  
885 caused by the activity of atmosphere monitoring instruments, mostly by lidar  
886 stations and the Central Laser Facility, readout of the electronics, and any  
887 hardware or software problems.

888 The value of uptime fraction has been derived from the data and is cross-  
889 checked by several techniques. The average uptime fraction for the whole  
890 observation period is around 13 % of the total time since operations commenced.  
891 Averaged uptime fractions for individual telescopes are shown in Fig. 29. For  
892 Los Leones and Coihueco the values refer to the period January 2005 to January  
893 2008, for Los Morados and Loma Amarilla the average is calculated from the  
894 individual start of operation till January 2008.

895 The Loma Amarilla building does not yet have a dedicated power line.  
896 The site is powered by a generator, which is less reliable and has caused a  
897 lower uptime compared to other FD sites. The knowledge of uptime fraction is  
898 essential for physics analysis such as hybrid spectrum determination, for details  
899 see [32].

### 900 *6.2. Background conditions*

901 The presence of the moon above the horizon increases the background light  
902 level, which has to be monitored. The direct current induced by the background  
903 is eliminated by the AC coupling of the PMT base, but it is possible to determine  
904 background levels using the direct relation between the fluctuations of the sky  
905 background and the photon flux. The analysis of the fluctuations in the ADC  
906 signal (variance analysis) performed for each night of data taking is used to  
907 monitor the FD background signal and data taking conditions. Thus the amount  
908 of light is derived from ADC variance value and it is given in units (ADC  
909 counts)<sup>2</sup>.

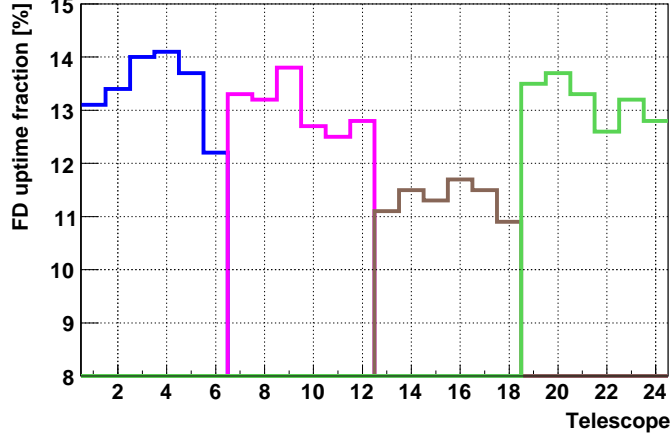


Figure 29: Uptime fraction for fluorescence telescopes. Telescopes are numbered as follows: Los Leones (south) site 1-6, Los Morados (east) site 7-12, Loma Amarilla (north) site 13-18, Coihueco (west) site 19-24.

910 The total background signal is the sum of the electronics background (pho-  
 911 tomultiplier and electronics noise) and the sky brightness (airglow, moonlight,  
 912 stars and planet light, zodiacal light, twilight and artificial light) [33]. The  
 913 typical values of background signals are: 3-5 (ADC counts)<sup>2</sup> for electronics  
 914 background, around 20 (ADC counts)<sup>2</sup> for cloudy nights, and between 25 and  
 915 60 (ADC counts)<sup>2</sup> for clear moonless nights. During nights when the moon is  
 916 above the horizon, the light background level can reach several hundred (ADC  
 917 counts)<sup>2</sup>.

918 The optimal background conditions for observation range from 25 to 60  
 919 (ADC counts)<sup>2</sup> which corresponds to photon background flux from approxi-  
 920 mately 100 to 250 photons m<sup>-2</sup>deg<sup>-2</sup>μs<sup>-1</sup>. Under such conditions, one event  
 921 per 2 hours is recorded for which the energy is determined with reconstruction  
 922 uncertainty smaller than 20% and the depth of maximum measured with an  
 923 uncertainty of better than 40 g/cm<sup>2</sup>. These high-quality events are the ones  
 924 used for physics analysis.

### 925 6.3. Standard Operation

926 The FD operation is not fully automated and at present the assistance of a  
 927 shift crew of at least two people per night is necessary. Their responsibilities  
 928 consist of several activities before, during and after measurement each night.  
 929 These include relative calibration of the cameras and optical components  
 930 before and after observation, starting and stopping data taking according to  
 931 weather conditions, prompt correction of software or hardware defects, etc. The  
 932 operation of the FD is still evolving, and the software development is steadily  
 933 transferring the responsibility of the human crew to automatic operation. The

934 ultimate aim is to operate FD telescopes in fully automatic and remotely  
935 supervised mode.

936 FD is operated in nights with moon fraction below 60 % beginning at the  
937 end of the astronomical twilight till the beginning of the next astronomical  
938 twilight. These criteria have evolved over time; before January 2005 the  
939 maximal illuminated moon fraction was 50%, and hence data taking period  
940 was 2 days shorter per month. The observation period lasts 16 days per month,  
941 with an average observational time of about 10 hours (ranging from about 14  
942 hours in June to 5 hours in December).

#### 943 *6.4. Slow Control System*

944 The fluorescence detectors are operated from the central campus in Malargüe,  
945 and are not operated directly from the FD buildings. Therefore, the main task  
946 of the Slow Control System (SCS) is to ensure a secure remote operation of the  
947 FD system. The SCS works autonomously and continuously monitors detector  
948 and weather conditions. Commands from the remote operator are accepted  
949 only if they do not violate safety rules that depend on the actual experimental  
950 conditions: high-voltage, wind speed, rain, light levels inside/outside the  
951 buildings, etc. In case of external problems such as power failures or  
952 communication breakdowns the SCS performs an orderly shutdown, and also  
953 a subsequent startup of the fluorescence detector system if the conditions have  
954 changed. If parts of the SCS itself fail, the system automatically reverts to a  
955 secure mode as all potentially critical system states (open shutters, high-voltage  
956 on, etc.) have to be actively maintained.

957 To ensure reliable supervision and to allow for high flexibility and stability,  
958 the SCS is based on industrial PROFIBUS components. This bus system  
959 consists of several bus terminals with specific functions, such as analog input,  
960 digital output, relays, etc. The terminals make up a modular bus-system which  
961 is addressed and controlled from a PC. The slow control PC is the central  
962 instance of the SCS in each FD building and runs the main control software  
963 under a Windows operating system. The system for one of the fluorescence  
964 detector buildings is sketched in Fig. 30.

965 The shift crew interacts with the control-system via web-browsers. One  
966 central webserver communicates via OPC-XML gateways with 4Control OPC-  
967 servers on the control-PCs. Several views with different levels of detail  
968 display the status, from an overview of the whole fluorescence detector of the  
969 Observatory down to single telescopes (see Fig. 31).

970 For the further automation of the data taking, an interface between the Slow  
971 Control and the Data Acquisition systems is under development. The final goal  
972 is an automatic, scheduled operation where the shifters have to react only to  
973 malfunctions of the system.

#### 974 *6.5. Monitoring*

975 FD data-taking can only take place under specific environmental conditions.  
976 The light-sensitive cameras must be operated on dark nights with low wind and



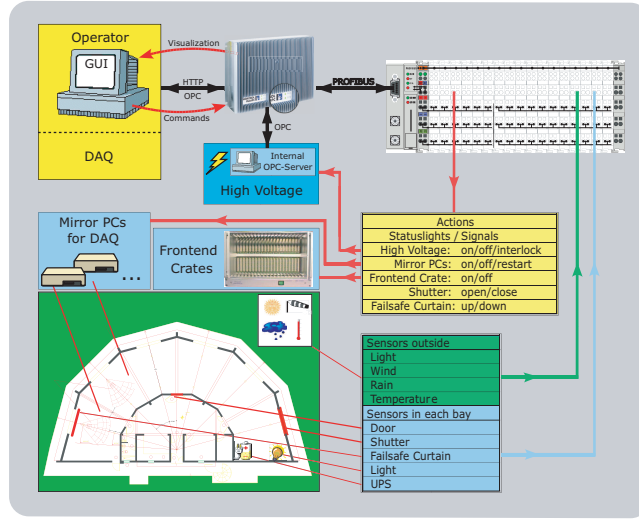


Figure 30: Schematic view of the slow-control system

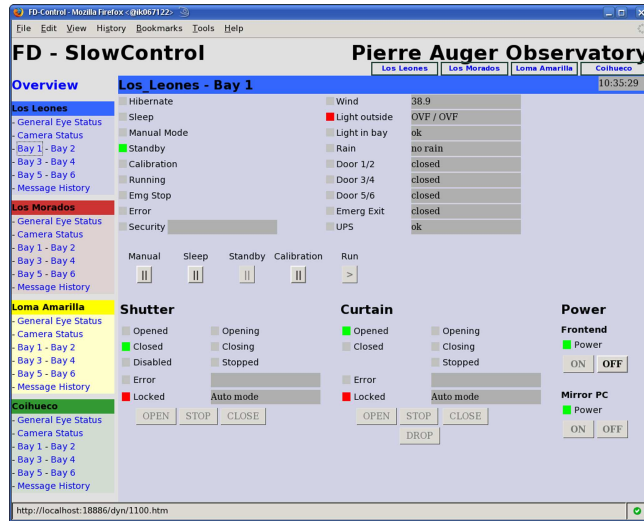


Figure 31: SCS display of a FD telescope

977 without rain. This makes the operation a full task for the shifters, who judge  
 978 the suitability of operations on the basis of the information given by the SCS.  
 979 The telescope performance must be monitored constantly to assure the quality  
 980 of the recorded data, as well as guarantee the safe operation of all detector  
 981 components. A user-friendly monitoring tool has been developed to support the

shifters in judging and supervising the status of the detector components, the electronics, and the data-acquisition.

The monitoring tool [45] is built upon online MySQL databases, organized by FD building. The databases record data from regular FD measurements and from calibrations and atmospheric surveys. The data are transferred to a central server at the Malargüe campus using the internal MySQL database replication mechanism. This mechanism recognizes communication problems and tries to synchronize database changes when the connection is reestablished; this guarantees the completeness of the dataset on the central server, even if the information are not available online immediately due to network failures. Fig. 32 shows the schematic layout of the databases.

The user interface is based on a webserver running Apache. The website uses PHP, CSS and JavaScript; graphs and custom visualizations are implemented using the JPGraph package for direct PHP calls to the database. They are dynamically generated to be accessible not only for the shifter, but also remotely for experts. Graph explanations as well as troubleshooting tips are available through linked wiki pages. Alarms are triggered automatically in case of special occurrences. This tool helps the shifter to monitor constantly the performance of the detector to assure the quality of the recorded data as well as guaranteeing the safe operation of all detector components. In addition it offers a unique opportunity to monitor the long term stability of some key quantities and the data quality.

#### 6.6. Online events

Besides using the online monitoring system for a continuous performance check by integrating over the most recent history of data, the operators are furnished with an online event display of the cameras. Two typical event examples corresponding to a real cosmic ray air shower and to a background event are shown in Fig. 34. The light spot of the air shower, shown in the upper part of the figure, takes about  $10 \mu\text{s}$  to cross the FOV of the camera while individual pixels are illuminated for up to about  $1 \mu\text{s}$ . This is clearly different

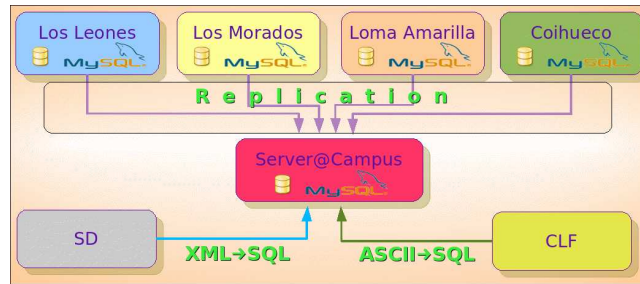


Figure 32: Organisation of the databases: The single databases at each FD-building are replicated to the database server at the central campus, while other sources like the SD insert data directly into the central database.

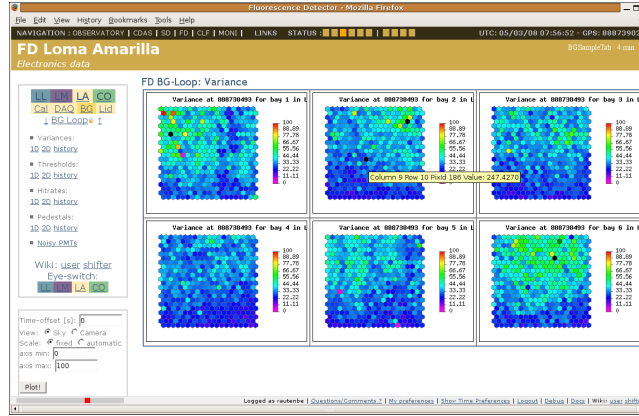


Figure 33: Screenshot of the FD monitoring web interface, showing a selection of FD background data for all pixels in the six cameras at Loma Amarilla.

from the event shown in the bottom part of the figure. Here, all activated pixels show huge narrow pulses all starting in the same sample of the 10 MHz ADC. Such a geometrical pixel pattern and timing feature is incompatible with an air shower and is most likely due to a muon penetrating the camera.

## 7. Event Reconstruction

### 7.1. Geometrical Reconstruction

A hybrid detector achieves the best geometrical accuracy by using timing information from all the detector components, both FD pixels and SD stations. Each element records a pulse of light from which it is possible to determine the time of the pulse and its uncertainty. Each trial geometry for the shower axis yields a prediction for the signal arrival times at each detector component.

Differences between actual and predicted times are weighted using their corresponding uncertainties, squared, and summed to construct a  $\chi^2$  value. The hypothesis with the minimum value of  $\chi^2$  is the reconstructed shower axis.

In the FD, cosmic ray showers are detected as a sequence of triggered pixels in the camera. An example of an event propagating through two adjacent FD telescopes is presented in Fig. 35. The first step in the analysis is the determination of the shower-detector plane (SDP). The SDP is the plane that includes the location of the eye and the line of the shower axis. (See the sketch in Fig. 36.) Experimentally, it is the plane through the eye which most nearly contains the pointing directions of the FD pixels centered on the shower axis. (See fitted line in Fig. 35.) Using a known axis provided from the Central Laser Facility (CLF), described in Ref. [14], the SDP reconstruction error can be evaluated by comparing the space angle between the normal vector to the experimentally determined SDP and the known true normal vector. This

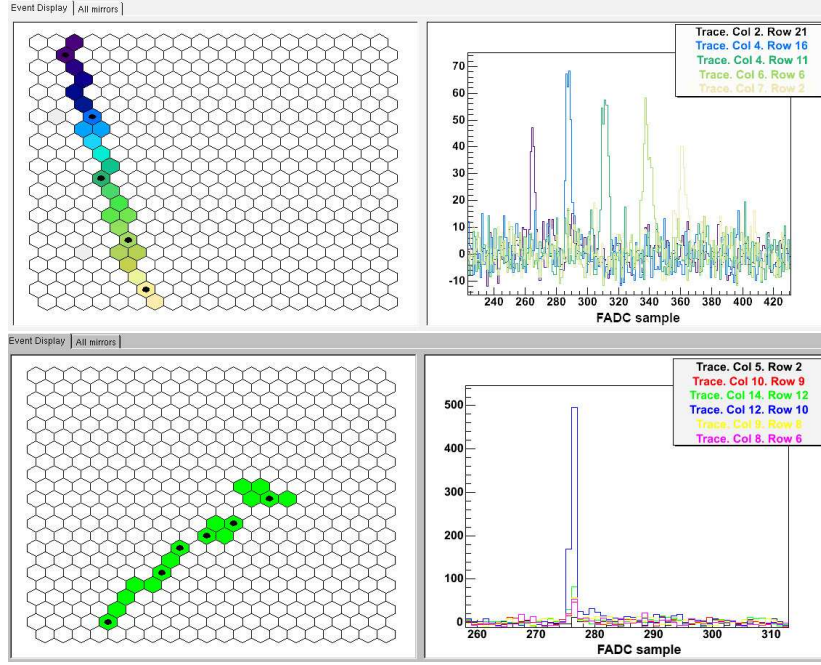


Figure 34: Top: A cosmic ray shower event as it appears in the event display. The pattern of the activated pixels is shown in the left panel while the right panel exhibits the response of the selected pixels as a function of time for the pixels marked by a black dot. The bin size is 100 ns. The development of the shower in the atmosphere can be qualitatively seen. Bottom: A background event most likely due to cosmic ray muon interacting with the glass of the PMT. All activated pixels give signals at the same time, a feature which is not compatible with a cosmic ray shower.

1037 uncertainty in the SDP is of the order of a few tenths of a degree depending on,  
 1038 for example, the length of the observed track in the camera.

1039 Next, the timing information of the pixels is used for reconstructing the  
 1040 shower axis within the SDP. As illustrated in Fig. 36, the shower axis can  
 1041 be characterized by two parameters: the perpendicular distance  $R_p$  from the  
 1042 camera to the track, and the angle  $\chi_0$  that the track makes with the horizontal  
 1043 line in the SDP. Each pixel which observes the track has a pointing direction  
 1044 which makes an angle  $\chi_i$  with the horizontal line. Let  $t_0$  be the time when the  
 1045 shower front on the axis passes the point of closest approach  $R_p$  to the camera.  
 1046 The light arrives at the  $i^{\text{th}}$  pixel at the time

$$t_i = t_0 + \frac{R_p}{c} \tan[(\chi_0 - \chi_i)/2]. \quad (1)$$

1047 The shower parameters are determined by fitting the data points to this  
 1048 functional form. Using the fast sampling electronics, such a monocular  
 1049 reconstruction may achieve excellent accuracy. However, the accuracy of the

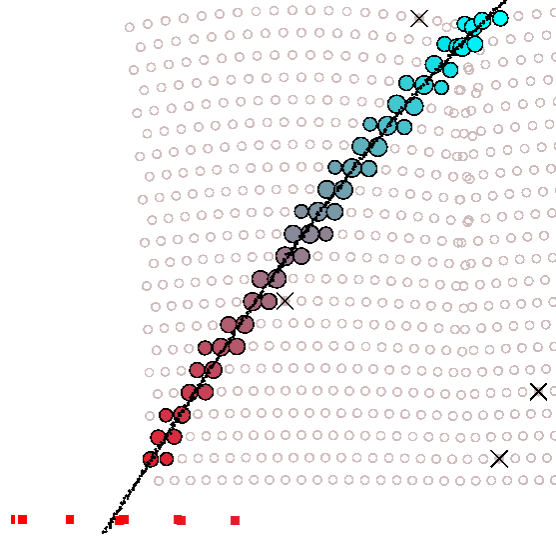


Figure 35: Light track of a hybrid event as seen by the fluorescence telescopes. The different colors indicate the timing sequence of the triggered pixels. The full line is the fitted shower-detector plane. (See text for explanation.) The red squares in the bottom left represent the surface stations that also triggered in this event. The crosses mark camera pixels that had a signal within the time of the trigger, but were marked by the reconstruction algorithm as too far either in distance (to the shower-detector plane) or in time (to the time fit).

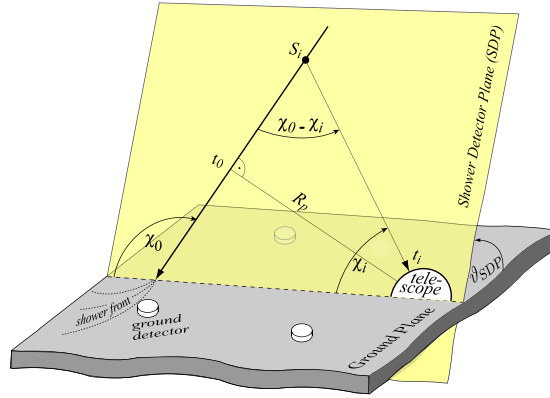


Figure 36: Illustration of the geometrical shower reconstruction from the observables of the fluorescence detector [50].

monocular reconstruction is limited when the measured angular speed  $d\chi/dt$  does not change much over the observed track length. An example is shown in Fig. 37. For these events (usually short tracks) there is a small curvature in the functional form of Eq. (1) such that there is a family of possible  $(R_p, \chi_0)$

axis solutions.  $R_p$  and  $\chi_0$  are tightly correlated, but neither value is well constrained. This leads to uncertainty in other shower parameters, including the reconstructed shower energy.

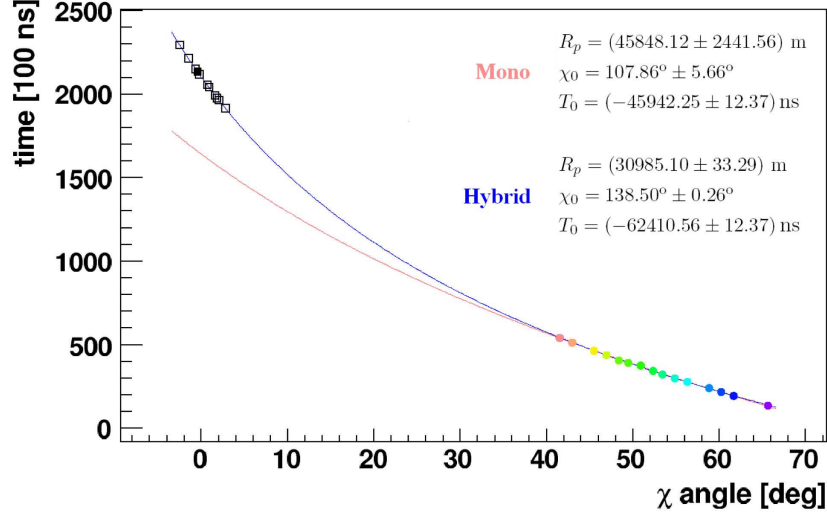


Figure 37: Functional form that correlates the time of arrival of the light at each pixel with the angle between the pointing direction of that particular pixel and the horizontal line within the shower-detector plane. FD data (color points) and SD data (squares) are superimposed to the monocular (red line) and hybrid (blue line) reconstruction fits. The full square indicates the SD station with the highest signal. This is a typical event in which the monocular reconstruction does not work well.

The fit degeneracy can be broken by combining the timing information from the SD stations with that of the FD telescopes. This is called the *hybrid* reconstruction. The hybrid solution for the example shown in Fig. 37 is shown as a blue line and the uncertainties in the parameters are specified in the legend.

Since the SD operates with a 100% duty cycle, most of the events observed by the FD are in fact hybrid events. There are also cases where the fluorescence detector, having a lower energy threshold, promotes a sub-threshold array trigger (see section 4.2.2). Surface stations are then matched by timing and location. This is an important capability because these sub-threshold hybrid events would not have triggered the array otherwise. In fact, the time of arrival at a single station at ground can suffice for the hybrid reconstruction.

The reconstruction uncertainties are validated using events with *known* geometries, i.e. light scattered from laser pulses. Since the location of the CLF (approximately equidistant from the first three fluorescence sites) and the direction of the laser beam are known to an accuracy better than the expected angular resolution of the fluorescence detector, laser shots from the CLF can be used to measure the accuracy of the geometrical reconstruction. Furthermore, the laser beam is split and part of the laser light is sent through an optical

1075 fiber to a nearby surface array station. Thus, the axis of the laser light can be  
 1076 reconstructed both in monocular mode and in the *single-tank* hybrid mode.

1077 Using the timing information from the telescope pixels together with the  
 1078 surface stations to reconstruct real air showers, a core location resolution of  
 1079 50 m is achieved. The typical resolution for the arrival direction of cosmic rays  
 1080 is  $0.6^\circ$  [49]. These results for the hybrid accuracy are in good agreement with  
 1081 estimations using analytic arguments [51], measurements on real data using a  
 1082 bootstrap method [52], and previous simulation studies [53].

### 1083 7.2. Shower Profile and Energy Reconstruction

1084 Once the geometry of the shower is known, the light collected at the aperture  
 1085 as a function of time can be converted to energy deposit at the shower as a  
 1086 function of slant depth. For this purpose, the light attenuation from the shower  
 1087 to the telescope needs to be estimated and all contributing light sources need  
 1088 to be disentangled [54]: fluorescence light [9, 11, 55, 56], direct and scattered  
 1089 Cherenkov light [57, 58] as well as multiple-scattered light [59, 60].  
 1090 An example of the measured light at the telescope aperture and the recon-  
 1091 structed light contributions and energy deposit profile is shown in Figs. 38  
 1092 and 39.

1093 The calorimetric energy of a shower is estimated by fitting a Gaisser-Hillas  
 1094 function [61] to the reconstructed energy deposit profile and integrating it.  
 1095 Finally, the total energy of the shower is obtained by correcting for the  
 1096 ‘invisible energy’ carried away by neutrinos and high energy muons. This  
 1097 correction is obtained from Monte-Carlo shower simulations [62] as the average  
 1098 correction factor for showers induced by different primary particles. However  
 1099 the differences between the correction factors for different primaries are on the  
 1100 level of a few percent only. After quality selection, the energy resolution (defined  
 1101 as event-to-event statistical uncertainty) of the fluorescence detector is 10% [63].  
 1102 The current systematic uncertainties on the energy scale sum up to 22%. The  
 1103 largest uncertainties are given by the absolute fluorescence yield (14%), the  
 1104 absolute calibration of the fluorescence telescopes (9%) and the uncertainty due  
 1105 to the reconstruction method of the longitudinal shower profile (10%).

### 1106 7.3. Detector Exposure

1107 The detection volume of fluorescence detectors varies with energy, as showers  
 1108 with higher energies emit more light and can be detected further away from the  
 1109 detector. The aperture also depends on environmental factors such as night-sky  
 1110 background light and atmospheric conditions. To model these effects, large sets  
 1111 of detailed MC simulations are used. The response of the Auger fluorescence  
 1112 telescopes has been simulated and the detector aperture has been estimated  
 1113 as a function of energy, average atmospheric conditions, and primary cosmic  
 1114 ray mass for a fixed configuration: a fully built detector with four fluorescence  
 1115 detectors and a 3000 km<sup>2</sup> surface array [46, 47]. The aperture increases from  
 1116 approximately 900 km<sup>2</sup> sr at  $10^{17.5}$  eV to about 7400 km<sup>2</sup> sr at  $10^{19}$  eV.

1117 However, the more important quantity here is the *hybrid detector exposure*  
 1118 which accounts for both the time variability of the atmospheric conditions and



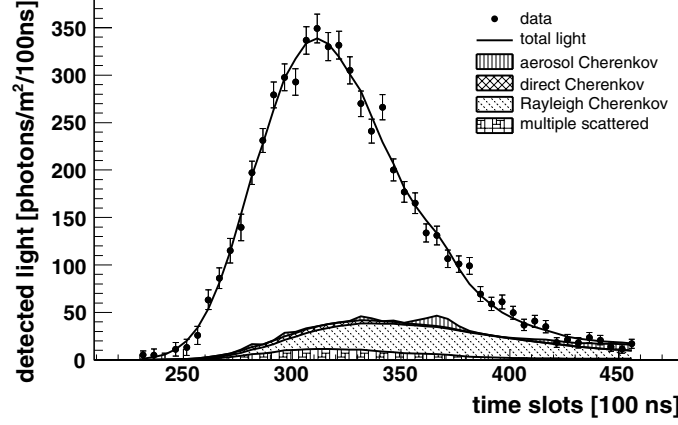


Figure 38: Example of a light-at-aperture measurement (dots) and reconstructed light sources (hatched areas).

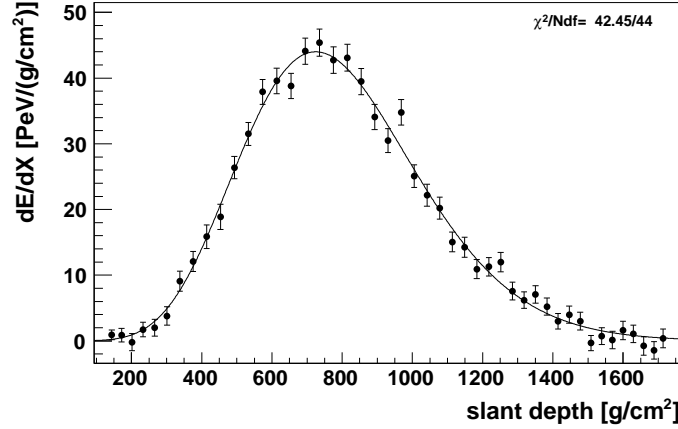


Figure 39: Energy deposit profile reconstructed from the light at aperture shown in Fig. 38. The line shows a Gaisser-Hillas fit of the profile. The energy reconstruction for this shower was  $3.0 \pm 0.2 \cdot 10^{19}$  eV.

1119 the detector setup. Very detailed time-dependent detector Monte Carlo simula-  
 1120 tions are used to reproduce the actual data taking conditions of all components  
 1121 of the Pierre Auger Observatory and to derive the hybrid exposure. To assure  
 1122 good agreement between data and Monte Carlo, extensive comparisons are  
 1123 performed at all reconstruction levels.

1124 To limit the influence of trigger threshold effects, an energy-dependent  
 1125 fiducial volume has been defined and strict quality criteria are applied during

the determination of the exposure [48]. For example, only data with a successful hybrid geometry reconstruction are selected, and the SD station used for the reconstruction has to lie within 750 m of the shower axis. This condition ensures that the probability to trigger at least one surface station is almost equal to one, and it significantly reduces the influence of the a-priori unknown primary mass composition. The resulting hybrid exposure, after all selection cuts accumulated during the 3 years of the building phase of the observatory, increases from about 50 km<sup>2</sup> sr yr at 10<sup>18</sup> eV to about 550 km<sup>2</sup> sr yr at 10<sup>19.5</sup> eV.

## 8. Summary

The Observatory has been in scientific operation since late 2003, and the synergy between the surface array and the fluorescence detector has proved to be most fruitful. In particular, the calorimetric FD energy measurements have provided a calibration for the high-statistics data set obtained with full-time operation of the surface array.<sup>11</sup>

This paper has described the Auger fluorescence detector system - its hardware, performance, calibration, and event reconstruction methods. The 24 wide-field telescopes view the atmosphere above the entire 3000-km<sup>2</sup> surface array.

Since the FD telescopes are operating well and the data is of high quality, there are no plans for major changes to the detectors. However, three additional telescopes known as HEAT [64], designed to view elevation angles from 30° to 58°, have recently begun operation at the Coihueco site. These telescopes will be used with the existing FD to study cosmic rays around and below 10<sup>18</sup>eV, in conjunction with the low energy extension to the SD known as AMIGA [65].

The Auger Observatory has opened a new chapter in cosmic ray physics because of both the large number of recorded high energy air showers and the quality of the hybrid measurements. The growing data set is expected to resolve the important questions relating to the highest energy particles of the universe.

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<sup>11</sup>Above the energy of the highest energy event measured by both fluorescence and surface detectors, the extrapolation of the same calibration curve is used to obtain the surface detector energies.

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